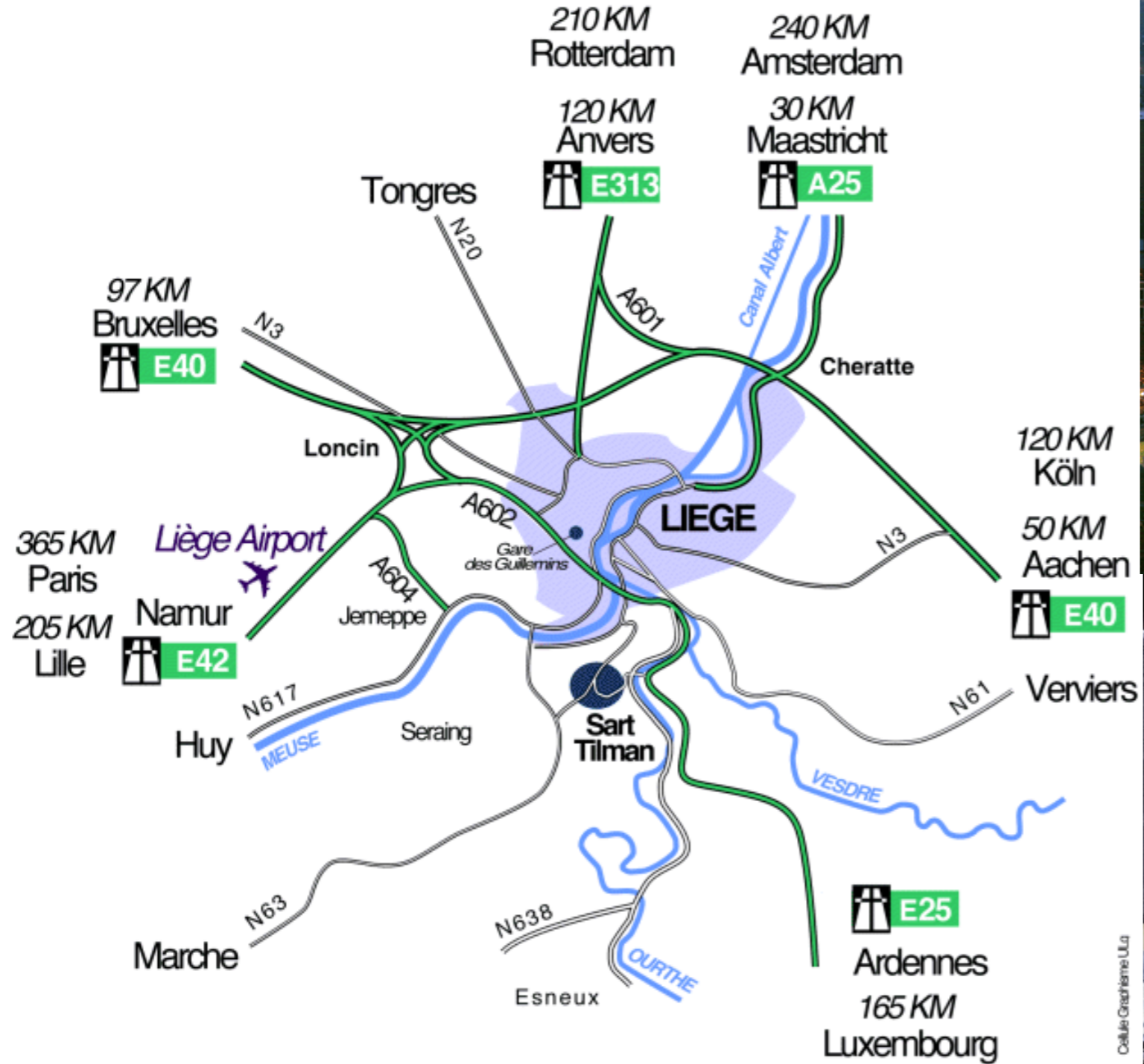


Testing of Materials under Space Radiation Environment Using Particle Accelerators of IPNAS

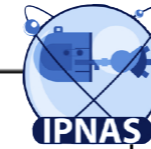
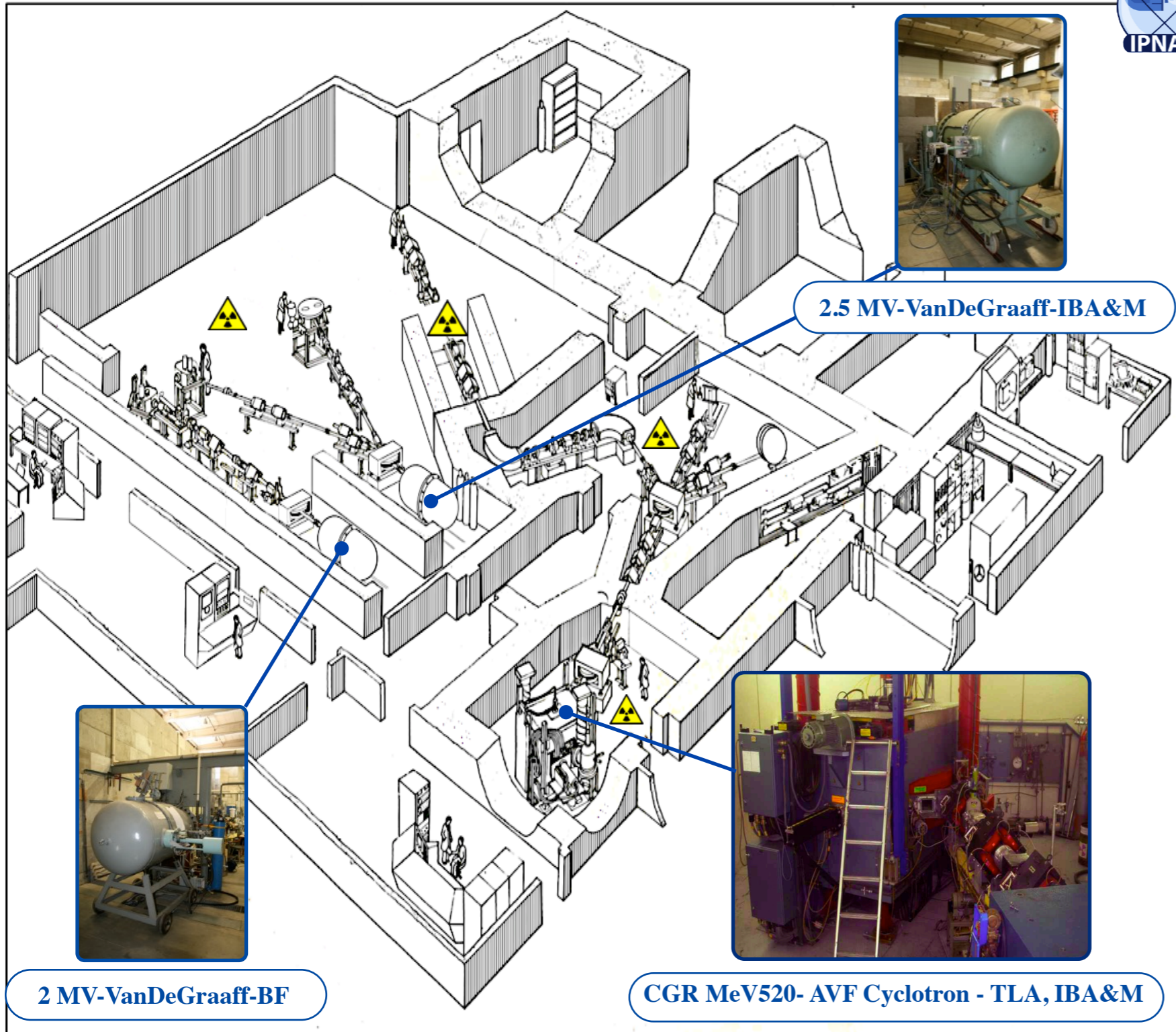
D. Strivay, L. Rossi, K. Fleury-Frenette, A. Carapelle, G. Chêne

June 16th 2017





- Fundamental physics:
 - Beam-foil spectroscopy
 - Cold atoms
 - Ionisation by charged particles and non-Rutherford diffusion cross section measurements
- Applied physics:
 - Ion Beam Analysis
 - Thin Layer Activation
 - Irradiation test
 - Cultural Heritage analysis



SANA Atomic and Nuclear Spectroscopy, Archaeometry

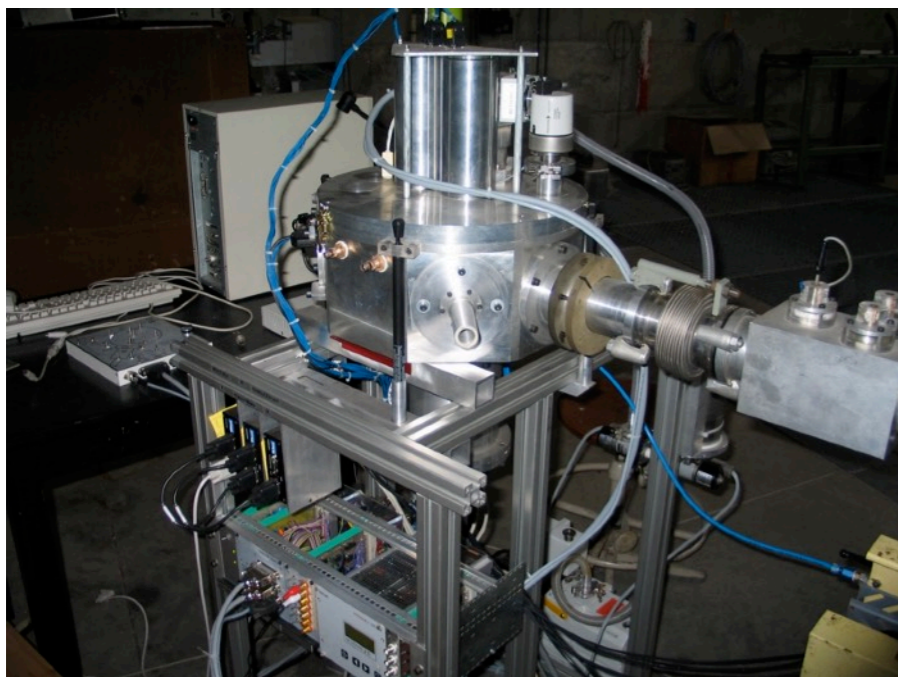
– Accelerators:

– 2 «oldies» Van De Graaff1 Cyclotron with variable energy

– Access to an extended **range of particle/energy** to **characterize/irradiate** materials: materials for energy production, Cultural Heritage objects, industrial applications (Thin Layer Activation)

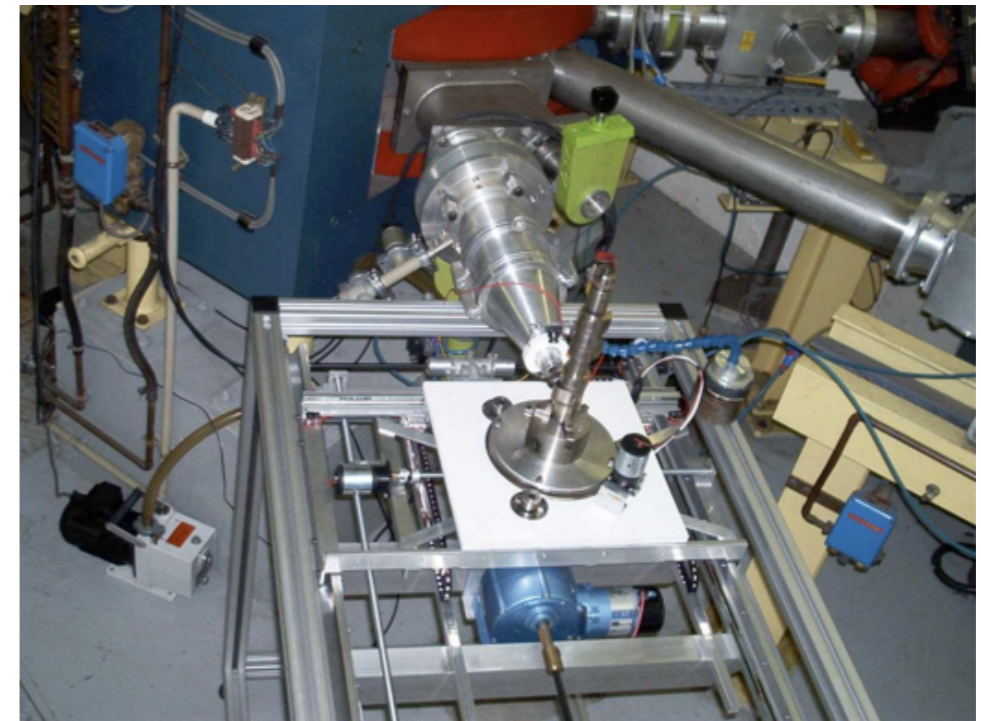
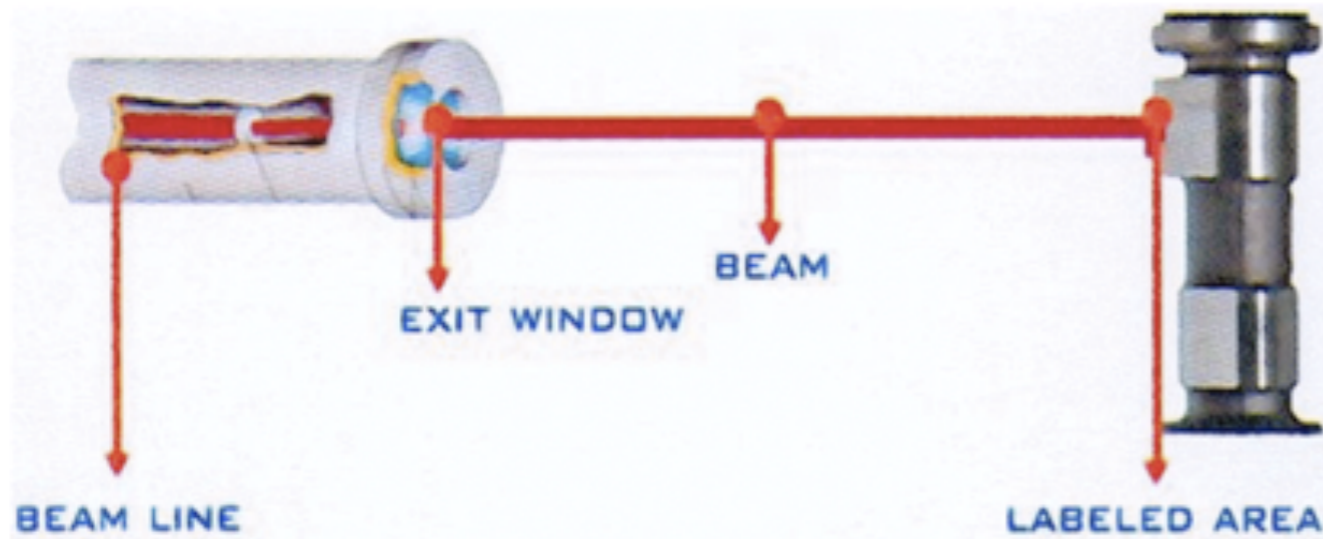
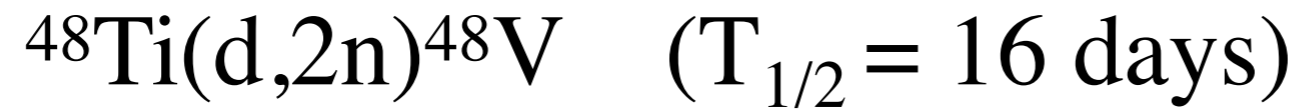
Accelerator	Particles	Energy range
VDG I - 2 MV	H ⁺ , D, He ⁺ , Metallic Ions, Li, Ne,	0.3- 2 MeV
VDG II - 2,5 MV	H ⁺ , D, ⁴ He ⁺ , ⁴ He ⁺⁺	0.3 - 2.5 MeV 0.3-4.5 MeV
CGR-MeV AVF Cyclotron	H ⁺ , D, ⁴ He ⁺ , ³ He ⁺	2.5- 22 MeV 4- 20MeV 5.5- 22 MeV

- Long history of IBA
- Applications in biology, thin films, polymers, geology, Cultural Heritage,....

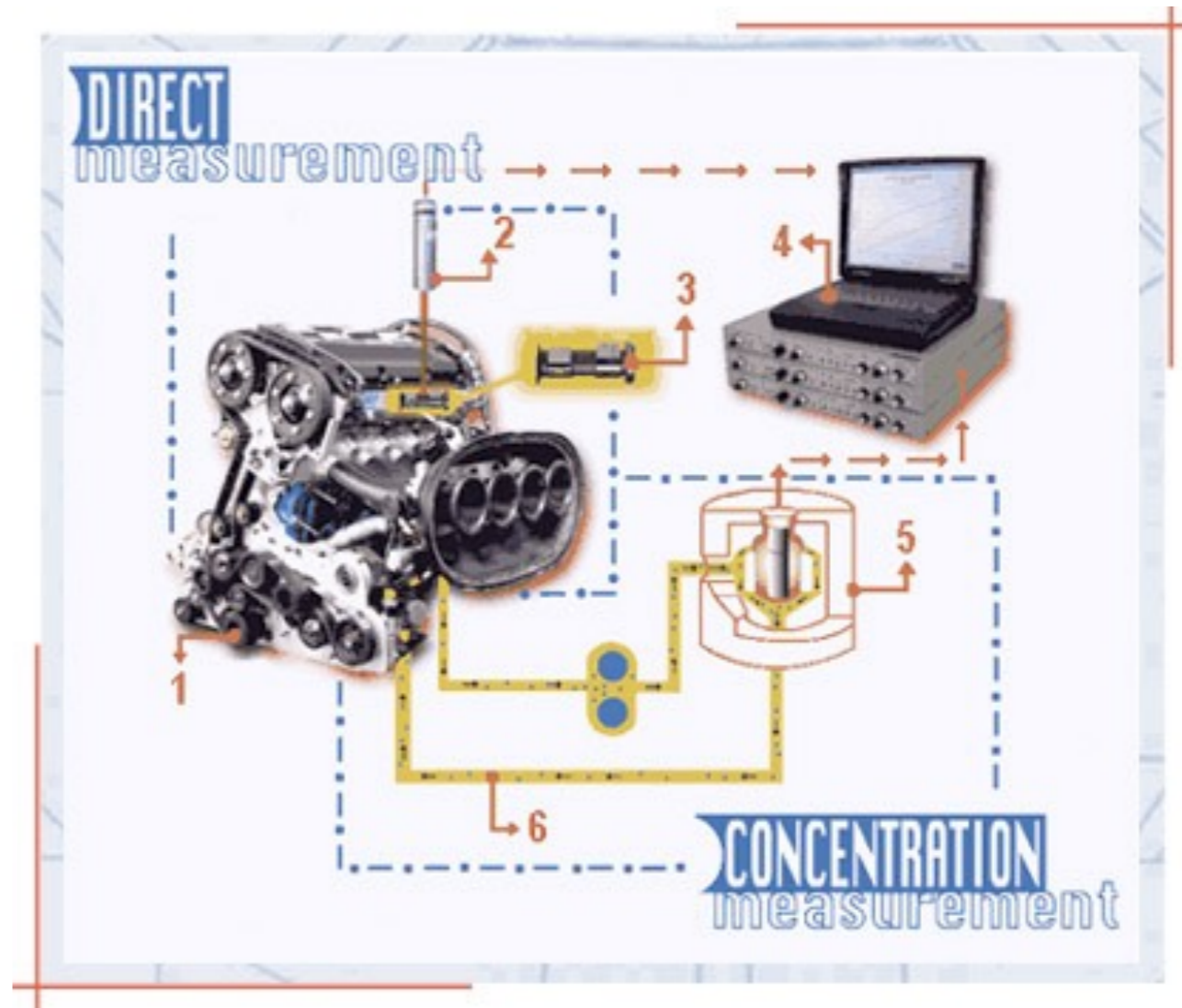


- ❖ 2 vacuum chambers on 2.5MV van de Graaff (RBS and multi-analysis)
- ❖ 4 beamlines on cyclotron (archaeometry, cross section measurements, irradiation, thin layer activation)

Typical reactions



- Direct measurement
(precision $\sim 0.1 \mu\text{m}$)
- Indirect measurement in the oil
(precision $\sim 0.01 \mu\text{m}$)



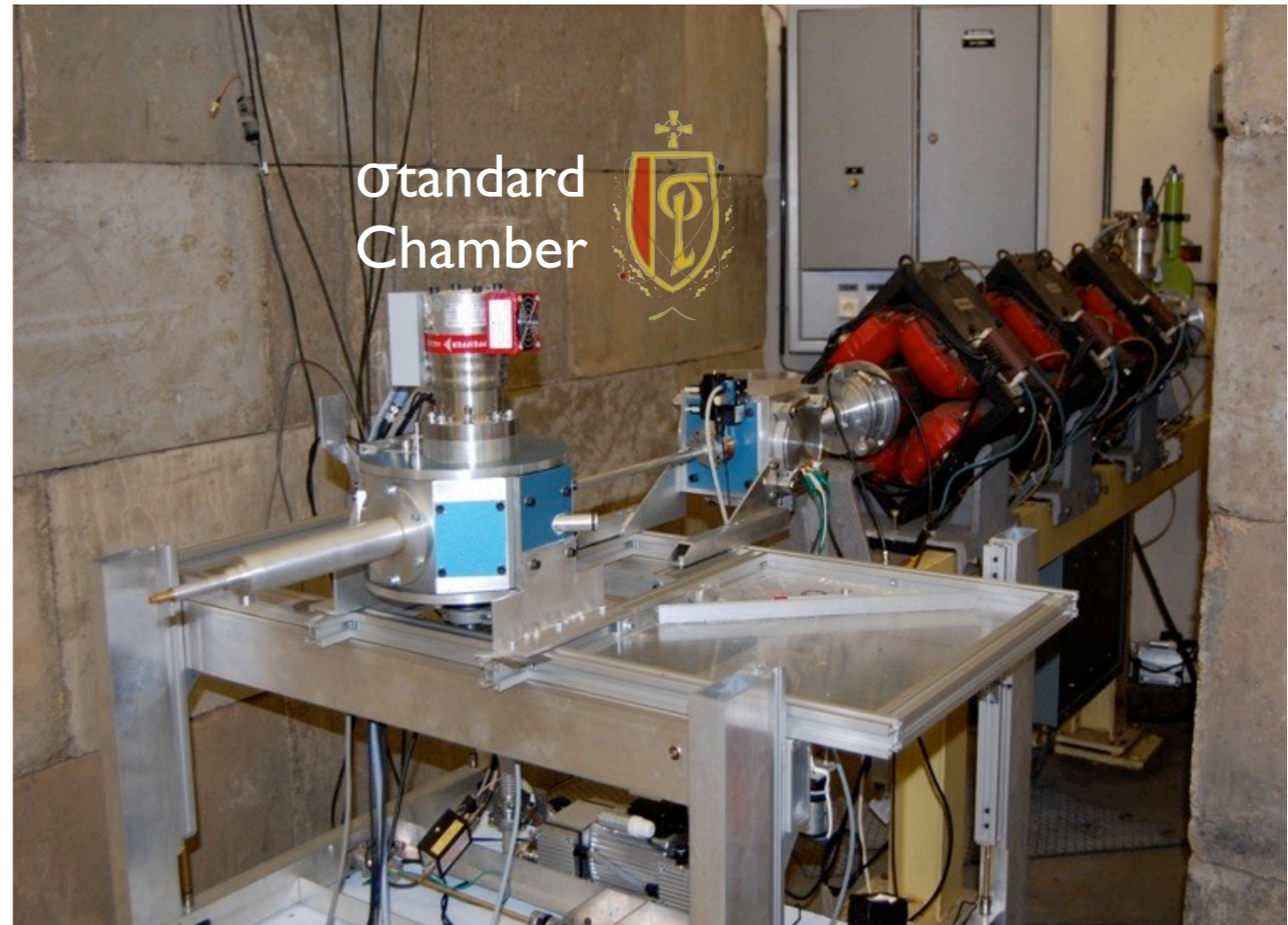
M. Scherge, K. Pölmann, A. Gervé, Wear **254** (2003) 801-817



- Funded in 2002, now part of the Research Unit Art, Archéologie et Patrimoine
- Analytical methods available: Ion Beam Analysis, SEM, Raman, FT-IR, XRF, IR reflectography, X-ray imaging, dendrochronology, 3D imaging...
- Expertise

- Analysis of Modern and Contemporary art paintings by Magritte, Picasso, Ensor, Chagall, Sam Francis, Otto Dix, Gauguin,...
- Roman wall-paintings
- Cave painting project (Font de Gaume)
- Medieval glass and tesserae production
- Modern pigment degradation
- Medieval manuscript degradation
- Polyurethane for outdoor sculptures

- Beamline for cross section measurements by 3-20 MeV proton and alpha
- Beam energy resolution of 1 keV
- Energy measured by TOF system
- X-ray production cross sections for high energy PIXE
- Non-Rutherford diffusion cross sections



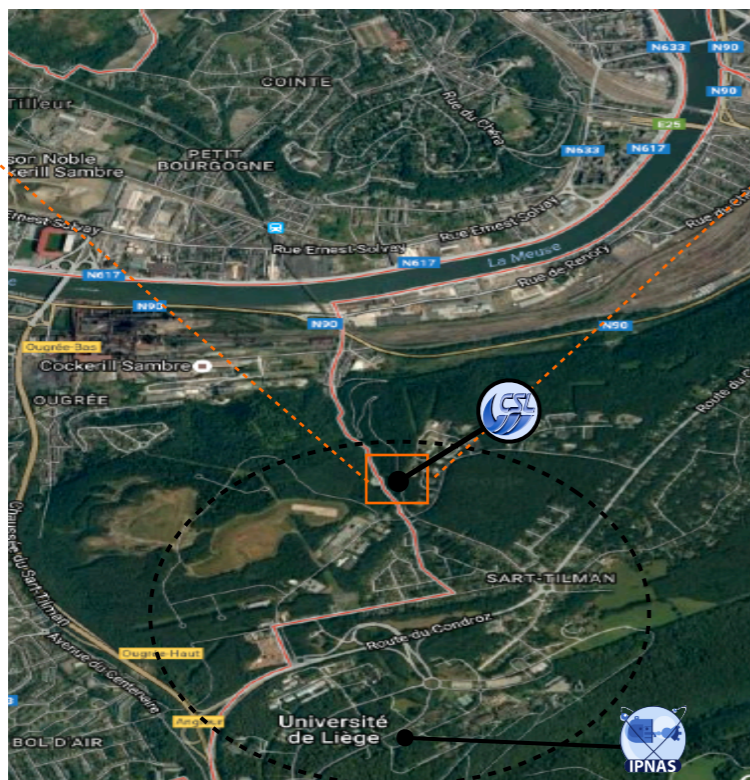


- **Centre Spatial de Liège (CSL)** : Applied research center (spin-off) from Liège University specialized in conception and testing of space instruments
- **Environment Testing for European Space Agency (ESA)**, space companies (**Thales, Airbus, OHB, RUAG...**) and regional firms (**AMOS, TechSpaceAero, TNO**)
- 100 scientists
- **Expertise** Electronics, Signal treatment, Metrology, Laser, Optics, Cryogeny, Vacuum

Laboratory of Micro&Nano Surface Engineering
Optical, thermal and mechanical tests under space environment

- **Equipements**

Coating techniques and surface Engineering system
Optical techniques surface characterization





- Centre Spatial de Liège (CSL) : Applied research center (spin-off) from Liège University specialized in conception and testing of space instruments

- **Environment Testing for European Space Agency (ESA)**, space companies (Thales, Airbus, OHB, RUAG...) and regional firms (AMOS, TechSpaceAero, TNO)

- 100 scientists

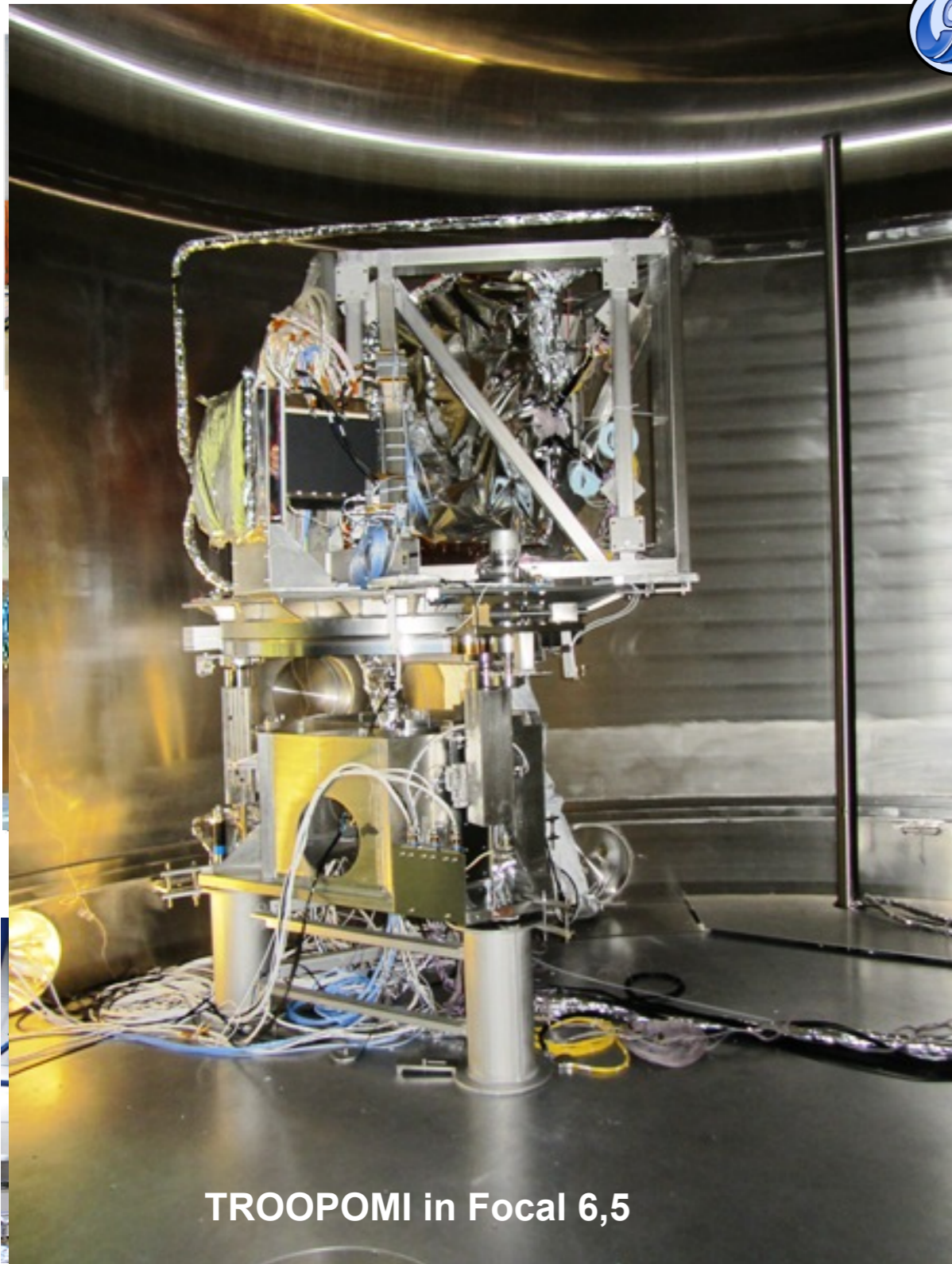
- **Expertise** Electronics, Signal treatment, Metrology, Laser, Optics, Cryogeny, Vacuum

Laboratory of Micro&Nano Surface Engineering
Optical, thermal and mechanical tests under space environment

- **Equipements**

Coating techniques and surface Engineering system
Optical techniques surface characterization





TROOPOMI in Focal 6,5

- Centre Spatial de Liège (CSL) : Applied research center (spin-off) from Liège University specialized in conception and testing of space instruments
- **Environment Testing for European Space Agency (ESA)**, space companies (**Thales, Airbus, OHB, RUAG...**) and regional firms (**AMOS, TechSpaceAero, TNO**)
- 100 scientists
- **Expertise** Electronics, Signal treatment, Metrology, Laser, Optics, Cryogeny, Vacuum

Laboratory of Micro&Nano Surface Engineering
Optical, thermal and mechanical tests under space environment

- **Equipements**
Coating techniques and surface Engineering system
Optical techniques surface characterization

IBA...

Quantitative analysis of materials:
thickness control, reproductibility, reverse engineering ... of optical coatings



IBMM...

Irradiation test platform :
H+ from 100 keV to 22 MeV
e- ?? ... conversion of VDG-I to produce e-beams

Aim : Produce charge particle beams characteristics of space radiation environment

Charge particles of interest

Accelerator	Particles	Energy range	Current range	Beam size
VDG I - 2 MV	H ⁺ , D, He ⁺ , Metallic Ions, Li, Ne,	0,5- 2 MeV	X	X
VDG II - 2,5 MV	H⁺, D, ⁴He⁺, ⁴ He ⁺⁺	0,1 - 2,5 MeV 0,5-4,5 MeV	0,1 nA - 1 μA	0,4 mm - 6 cm
CYCLOTRON AVF	H⁺, D, ⁴ He ⁺⁺ , ³ He ⁺⁺	2,5- 22 MeV 4- 20MeV 5,5- 22 MeV	0,3 nA - 100 μA	0,8 mm - 12 cm

Primary radiation environment : *Boudenot et al. RADECS 1999*

Provenance	Particles	Energy range	Flux
	H ⁺ e-	<qq. 100 MeV (dont 99% <10MeV) <7 MeV (dont 99% <2 MeV)	10 - 10 ⁶ cm ⁻² s ⁻¹ 10 ⁻² - 10 ⁷ cm ⁻² s ⁻¹
	H ⁺ , e- ⁴ He ⁺⁺	<100 keV <qq keV	10 ⁸ - 10 ¹⁰ cm ⁻² s ⁻¹
	H ⁺ , ⁴ He ⁺⁺ ions lourds	10 MeV- 1 GeV 10 MeV à qq 100 MeV	10 ¹⁰ cm ⁻² s ⁻¹ ~10 ² - 10 ³ cm ⁻² s ⁻¹
	H ⁺ , 87% ⁴ He ⁺⁺ 12% IONS LOURDS 1%	10 ² - 10 ⁶ MeV Fortes énergies 1-10 ¹⁴ MeV	1 cm ⁻² s ⁻¹ 100 MeV 10 ⁻⁴ cm ⁻² s ⁻¹ 10 ⁶ MeV

Challenge: how to attain these objectives with QA, ISO criteria used by Space agencies

IBA...

Quantitative analysis of materials:
thickness control, reproductibility, reverse engineering ... of optical coatings



IBMM...

Irradiation test platform :
H+ from 100 keV to 22 MeV
e- ?? ... conversion of VDG-I to produce e- beams

Aim : Produce charge particle beams characteristics of space radiation environment

Charge particles of interest

Accelerator	Particles	Energy range	Current range	Beam size
VDG I - 2 MV	H ⁺ , D, He ⁺ , Metallic Ions, Li, Ne,	0,5- 2 MeV	X	X
VDG II - 2,5 MV	H⁺, D, ⁴He⁺,	0,1 - 2,5 MeV	0,1 nA - 1 μA	0,4 mm - 6 cm
	⁴ He ⁺⁺	0,5-4,5 MeV		
CGR-MeV AVF Cyclotron	H⁺,	2,5- 22 MeV	0,3 nA - 100 μA	0,8 mm - 12 cm
	D,	4- 20MeV		
	⁴ He ⁺⁺ , ³ He ⁺⁺	5,5- 22 MeV		

Primary radiation environment : *Boudenot et al. RADECS 1999*

Provenance	Particles	Energy range	Flux
Radiation ring 250km < H < 80 000 km	H ⁺	<qq. 100 MeV (dont 99% <10MeV)	10 - 10 ⁶ cm ⁻² s ⁻¹
	e-	<7 MeV (dont 99% <2 MeV)	10 ⁻² - 10 ⁷ cm ⁻² s ⁻¹
Solar wind	H ⁺ , e- ⁴ He ⁺⁺	<100 keV <qq keV	10 ⁸ - 10 ¹⁰ cm ⁻² s ⁻¹
Cyclic solar eruption (10 years)	H ⁺ , ⁴ He ⁺⁺ ions lourds	10 MeV- 1 GeV	10 ¹⁰ cm ⁻² s ⁻¹
		10 MeV à qq 100 MeV	~10 ² - 10 ³ cm ⁻² s ⁻¹
Cosmic ray	H ⁺ , 87% ⁴ He ⁺⁺ 12% IONS LOURDS 1%	10 ² - 10 ⁶ MeV	1 cm ⁻² s ⁻¹ 100 MeV
		Fortes énergies	10 ⁻⁴ cm ⁻² s ⁻¹ 10 ⁶ MeV
		1-10 ¹⁴ MeV	

Challenge: how to attain these objectives with QA, ISO criteria used by Space agencies

IBA...

Quantitative analysis of materials:
thickness control, reproductibility, reverse engineering ... of optical coatings



IBMM...

Irradiation test platform :
H+ from 100 keV to 22 MeV
e- ?? ... conversion of VDG-I to produce e- beams

Aim : Produce charge particle beams characteristics of space radiation environment

Charge particles of interest

Accelerator	Particles	Energy range	Current range	Beam size
VDG I - 2 MV	H ⁺ , D, He ⁺ , Metallic Ions, Li, Ne,	0,5- 2 MeV	X	X
VDG II - 2,5 MV	H⁺, D, ⁴He⁺,	0,1 - 2,5 MeV	0,1 nA - 1 μA	0,4 mm - 6 cm
	⁴ He ⁺⁺	0,5-4,5 MeV		
CGR-MeV AVF Cyclotron	H⁺,	2,5- 22 MeV	0,3 nA - 100 μA	0,8 mm - 12 cm
	D,	4- 20MeV		
	⁴ He ⁺⁺ , ³ He ⁺⁺	5,5- 22 MeV		

Altitude & mission duration

Primary radiation environment : *Boudenot et al. RADECS 1999*

Provenance	Particles	Energy range	Flux	
Radiation ring 250km < H < 80 000 km	H ⁺	<qq. 100 MeV (dont 99% <10MeV)	10 - 10 ⁶ cm ⁻² s ⁻¹	
	e ⁻	<7 MeV (dont 99% <2 MeV)	10 ⁻² - 10 ⁷ cm ⁻² s ⁻¹	
Solar wind	H ⁺ , e ⁻ ⁴ He ⁺⁺	<100 keV <qq keV	10 ⁸ - 10 ¹⁰ cm ⁻² s ⁻¹	
Cyclic solar eruption (10 years)	H ⁺ , ⁴ He ⁺⁺ <small>ions lourds</small>	10 MeV- 1 GeV	10 ¹⁰ cm ⁻² s ⁻¹	
		10 MeV à qq 100 MeV	~10 ² - 10 ³ cm ⁻² s ⁻¹	
Orbite	Duration	Altitude	Inclinaison	Longitude λ
LEO (Low Earth Orbit)	7 ans	800 km	60°	λ (t)
GEO (GEOstationnaire)	18 ans	35786 km	0°	36°

Challenge: how to attain these objectives with QA, ISO criteria used by Space agencies

Contamination sources defined in «Space product assurance - ECSS-Q-ST-70-05C»

- Back-streaming from pumping 🕊️
- Outgassing products during vacuum integration 🕊️
- Cleaning products residues 🕊️
- Handling residues (e.g. human grease). 🕊️
- Unfiltered external pollution 😞

Table D-1: Assignment of infrared absorption bands for the four main groups of contaminants

Type of contaminant	Characteristic wave number (cm ⁻¹)	Functional group	Signal strength ^a	Remarks
Hydrocarbons	3000 - 2850	Alkanes (CH, CH ₂ , CH ₃)	s	2 or 3 bands, stretching
	3100 - 3020	Alkenes	m	Stretching
	1470 - 1440	-CH ₃	ms	Asymmetric deformation
	1390 - 1370	-CH ₃	m	Symmetric deformation
Esters	1750 - 1735	C=O	s	Stretching (saturated ester)
	1300 - 1050	C-O	s	Stretching
Methyl silicones	1280 - 1255	Si-CH ₃	vs	CH ₃ deformation
	1130 - 1000	Si-O-Si	s	Asymmetric stretching
	860 - 760	Si-CH ₃	vs	Si-C stretching or CH ₃ rocking ^b
Methyl phenyl silicones	1280 - 1255	Si-CH ₃	vs	CH ₃ deformation
	1130 - 1000	Si-O-Si	s	Asymmetric stretching
	1125 - 1100	Si-Aryl	vs	
	860 - 760	Si-CH ₃	vs	Si-C stretching or CH ₃ rocking ^b

^a Strength of signal: vs = very strong, s = strong, ms = medium to strong, m = medium.

^b One methyl: 765 cm⁻¹; two methyls: 855 cm⁻¹ and 800 cm⁻¹; three methyls: 840 cm⁻¹ and 765 cm⁻¹.

- **New pumping groups**
- **Integration of cold traps in the vacuum chambers**
- **New connections and vacuum feedthroughs**
- **Systematic cleaning procedure at each opening**
- **Use of contamination witnesses before and during irradiation and analysis by FTIR (labo QA - CSL) following protocol described in «Space product assurance - ECSS-Q-ST-70-05C»**



Contamination sources defined in «Space product assurance - ECSS-Q-ST-70-05C»

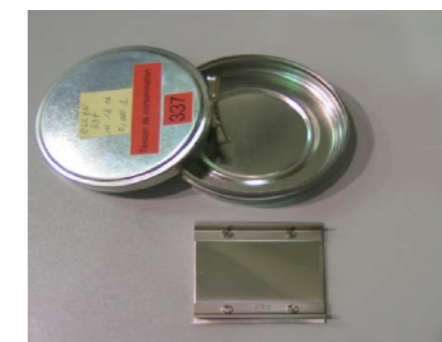
- Back-streaming from pumping 🙌
- Outgassing products during vacuum integration 🙌
- Cleaning products residues 🙌
- Handling residues (e.g. human grease). 🙌
- Unfiltered external pollution 😞

Table D-1: Assignment of infrared absorption bands for the four main groups of contaminants

Type of contaminant	Characteristic wave number (cm ⁻¹)	Functional group	Signal strength ^a	Remarks
Hydrocarbons	3000 - 2850	Alkanes (CH, CH ₂ , CH ₃)	s	2 or 3 bands, stretching
	3100 - 3020	Alkenes	m	Stretching
	1470 - 1440	-CH ₃	ms	Asymmetric deformation
	1390 - 1370	-CH ₃	m	Symmetric deformation
Esters	1750 - 1735	C=O	s	Stretching (saturated ester)
	1300 - 1050	C-O	s	Stretching
Methyl silicones	1280 - 1255	Si-CH ₃	vs	CH ₃ deformation
	1130 - 1000	Si-O-Si	s	Asymmetric stretching
	860 - 760	Si-CH ₃	vs	Si-C stretching or CH ₃ rocking ^b
Methyl phenyl silicones	1280 - 1255	Si-CH ₃	vs	CH ₃ deformation
	1130 - 1000	Si-O-Si	s	Asymmetric stretching
	1125 - 1100	Si-Aryl	vs	
	860 - 760	Si-CH ₃	vs	Si-C stretching or CH ₃ rocking ^b

^a Strength of signal: vs = very strong, s = strong, ms = medium to strong, m = medium.
^b One methyl: 765 cm⁻¹; two methyls: 855 cm⁻¹ and 800 cm⁻¹; three methyls: 840 cm⁻¹ and 765 cm⁻¹.

- **New pumping groups**
- **Integration of cold traps in the vacuum chambers**
- **New connections and vacuum feedthroughs**
- **Systematic cleaning procedure at each opening**
- **Use of contamination witnesses before and during irradiation and analysis by FTIR (labo QA - CSL) following protocol described in «Space product assurance - ECSS-Q-ST-70-05C»**



Contamination sources defined in «Space product assurance - ECSS-Q-ST-70-05C»

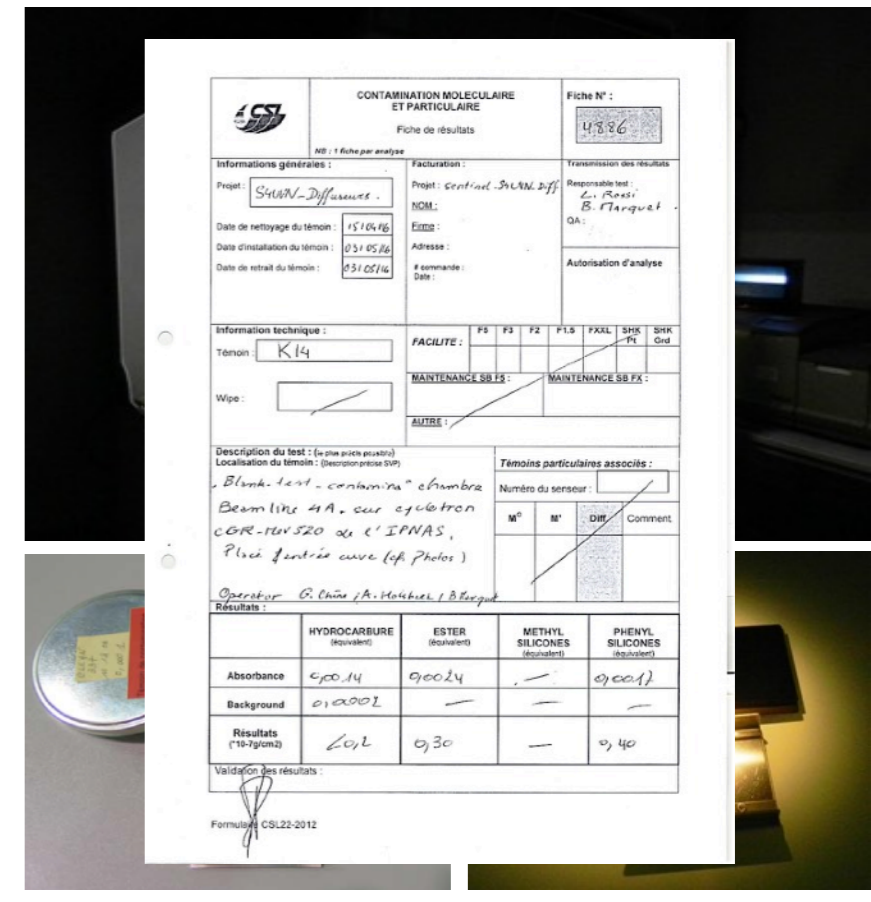
- Back-streaming from pumping 🙌
- Outgassing products during vacuum integration 🙌
- Cleaning products residues 🙌
- Handling residues (e.g. human grease). 🙌
- Unfiltered external pollution 😞

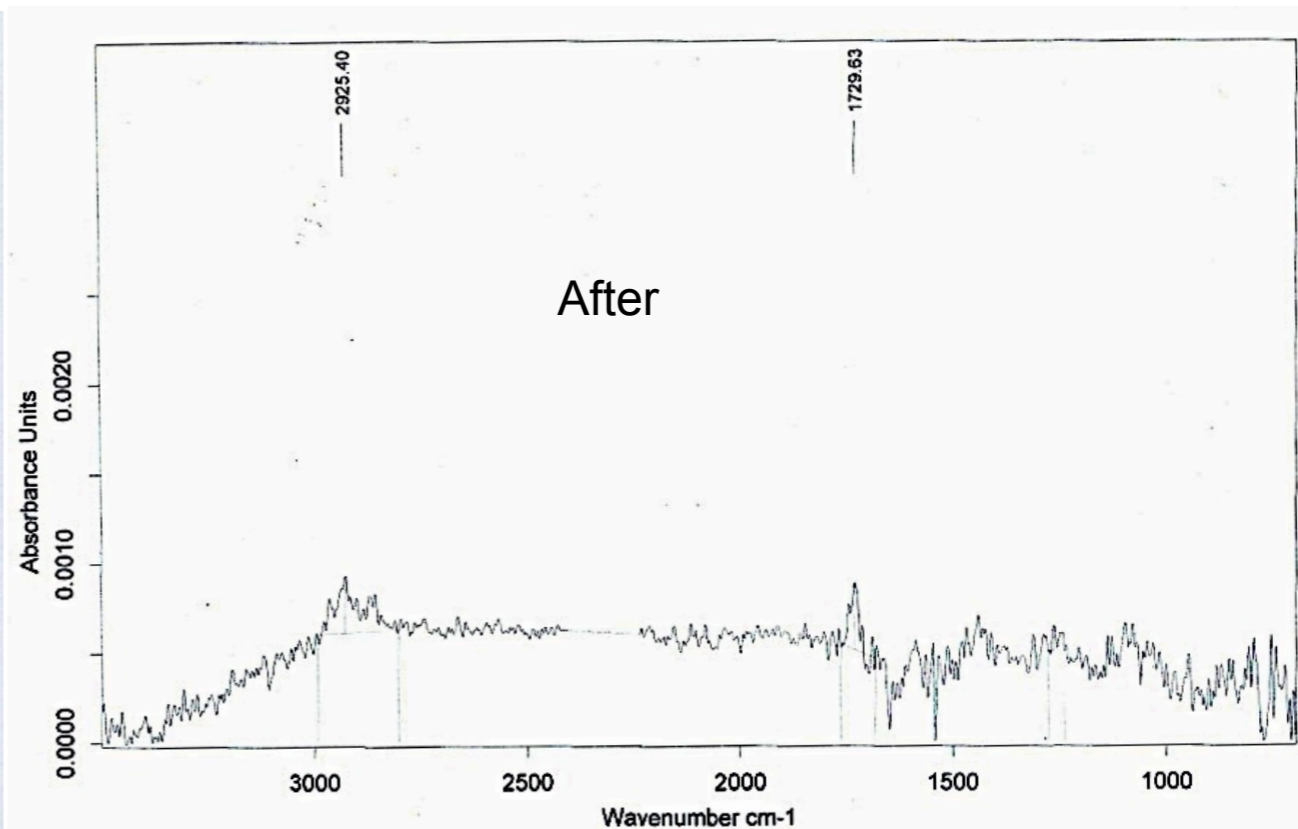
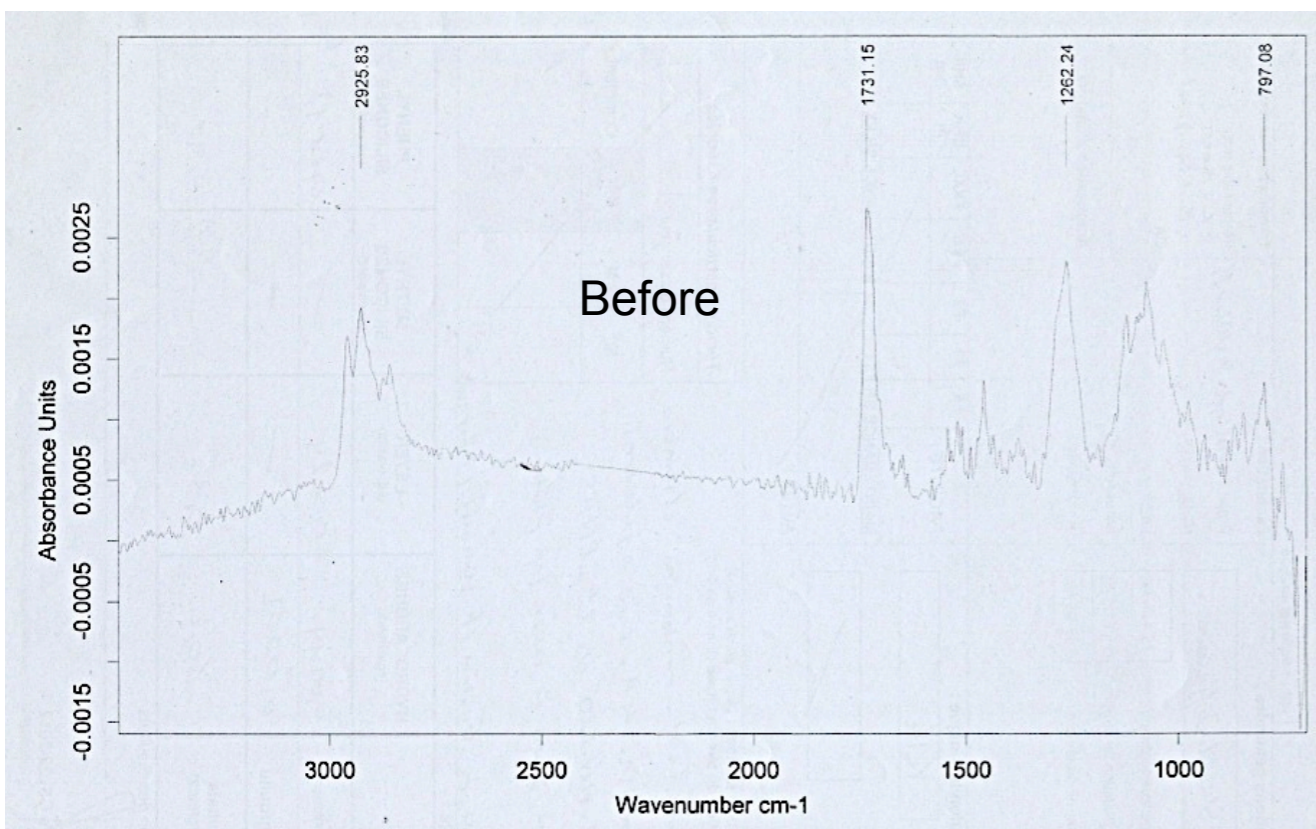
Table D-1: Assignment of infrared absorption bands for the four main groups of contaminants

Type of contaminant	Characteristic wave number (cm ⁻¹)	Functional group	Signal strength ^a	Remarks
Hydrocarbons	3000 - 2850	Alkanes (CH, CH ₂ , CH ₃)	s	2 or 3 bands, stretching
	3100 - 3020	Alkenes	m	Stretching
	1470 - 1440	-CH ₃	ms	Asymmetric deformation
	1390 - 1370	-CH ₃	m	Symmetric deformation
Esters	1750 - 1735	C=O	s	Stretching (saturated ester)
	1300 - 1050	C-O	s	Stretching
Methyl silicones	1280 - 1255	Si-CH ₃	vs	CH ₃ deformation
	1130 - 1000	Si-O-Si	s	Asymmetric stretching
	860 - 760	Si-CH ₃	vs	Si-C stretching or CH ₃ rocking ^b
Methyl phenyl silicones	1280 - 1255	Si-CH ₃	vs	CH ₃ deformation
	1130 - 1000	Si-O-Si	s	Asymmetric stretching
	1125 - 1100	Si-Aryl	vs	
	860 - 760	Si-CH ₃	vs	Si-C stretching or CH ₃ rocking ^b

^a Strength of signal: vs = very strong, s = strong, ms = medium to strong, m = medium.
^b One methyl: 765 cm⁻¹; two methyls: 855 cm⁻¹ and 800 cm⁻¹; three methyls: 840 cm⁻¹ and 765 cm⁻¹.

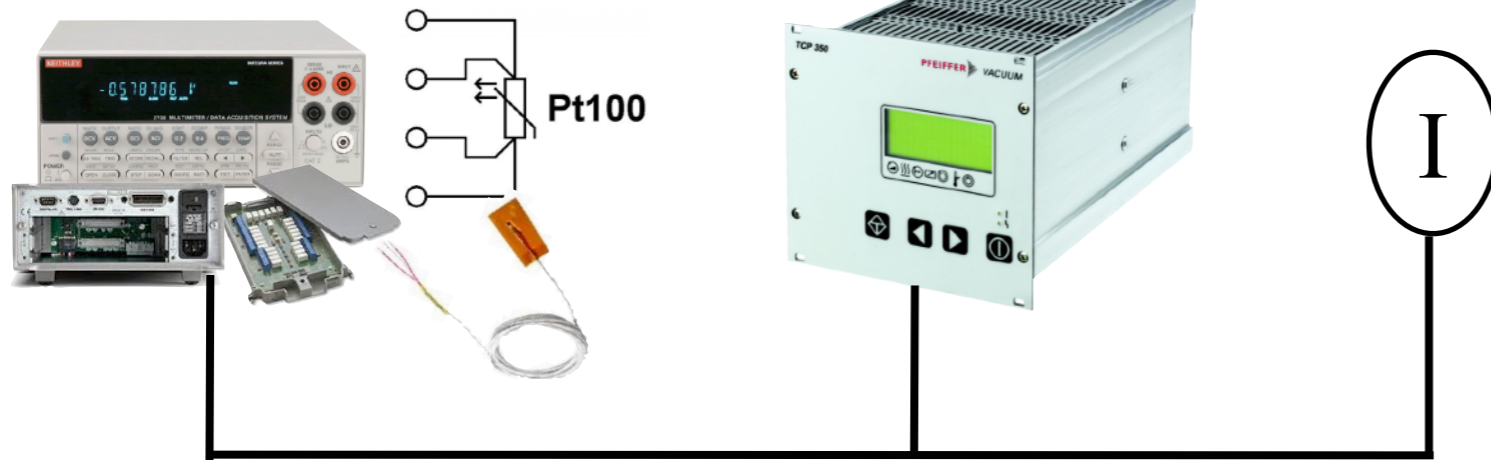
- **New pumping groups**
- **Integration of cold traps in the vacuum chambers**
- **New connections and vacuum feedthroughs**
- **Systematic cleaning procedure at each opening**
- **Use of contamination witnesses before and during irradiation and analysis by FTIR (labo QA - CSL) following protocol described in «Space product assurance - ECSS-Q-ST-70-05C»**





	HYDROCARBURE (2925 cm ⁻¹)	ESTER (1735 cm ⁻¹)	METHYL SILICONES	PHENYL SILICONES
Absorbance	0,0014	0,0024	/	0,0017
Background	0,0001	/	/	/
Results (x 10⁻⁷ g.cm²)	< 0,2	0,3	/	0,4
Absorbance	0,0003	0,0004	/	/
Background	0,0001	/	/	/
Results (x 10⁻⁷ g.cm²)	< 0,2	/	/	/

Thermal bridge between cold trap/
sample holder and heating system

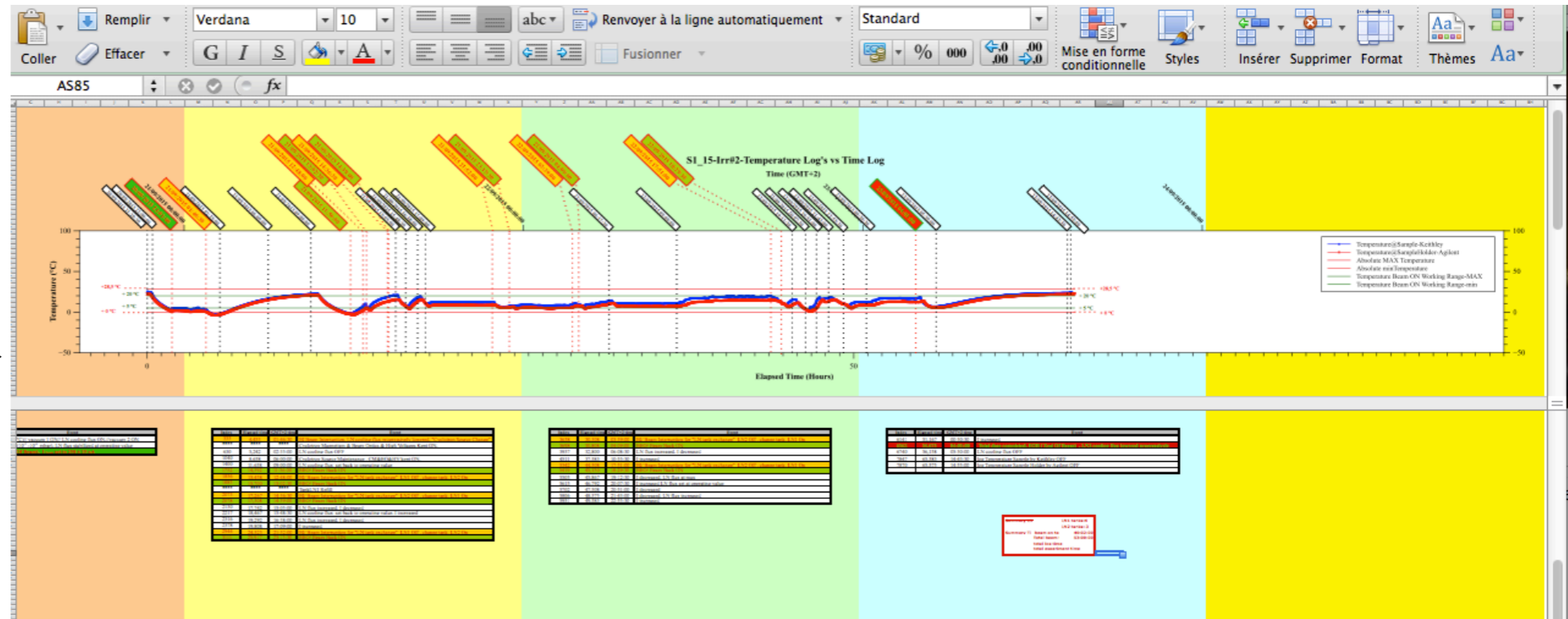
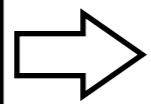


Standards CSL

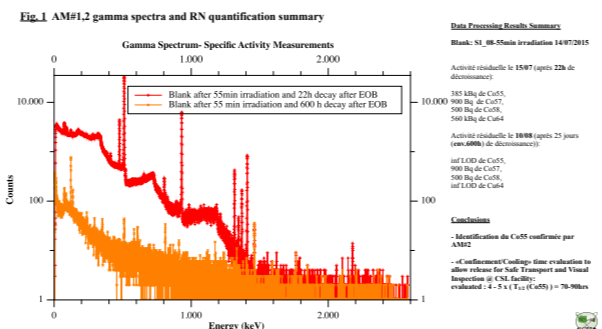
- T,P measurement and T control
- Low beam current (nA) and total dose measurements (<5%)
- Logs files .txt
- Deliverable .xls

```

date;hour;P;T1;T2;I
date;hour;P;T1;T2;I
date;hour;P;T1;T2;I
date;hour;P;T1;T2;I
date;hour;P;T1;T2;I
    
```



cyclotron «CSL Beam Line»
 H+ @ 10 MeV (S4UVN) , 20 MeV (Sider)
 use of Al foil as beam homogeneizer
 «CSL Cyclo Chamber»
 LN Cold Trap & I, T, P control and log (S4UVN)



Centre Spatial d

Figure 2: Ulg-IPNAS Van de Graaff vacuum chamber

Table 8-2: Measurement tools

	Description	Type	Type no	Accuracy
Current measurement	The current is measured directly on the chamber (-06). This current is integrated by a Digital Current Integrator Model 230 (Ortec). This device produces a digital pulse for each collected charge (1 pulse = 3e-10 C).	Ortec digital current integrator	439	5%
Temperature measurement	Two thermal sensors (PT100) (one nominal and one redundant) will be accommodated on the sample; One Keithley will read the PT100 and send the data to PC.	PT100 2791 Ethernet millimeter		±/-1°C
Pressure Measurement		Pfeiffer Vacuum compact full range gauge		

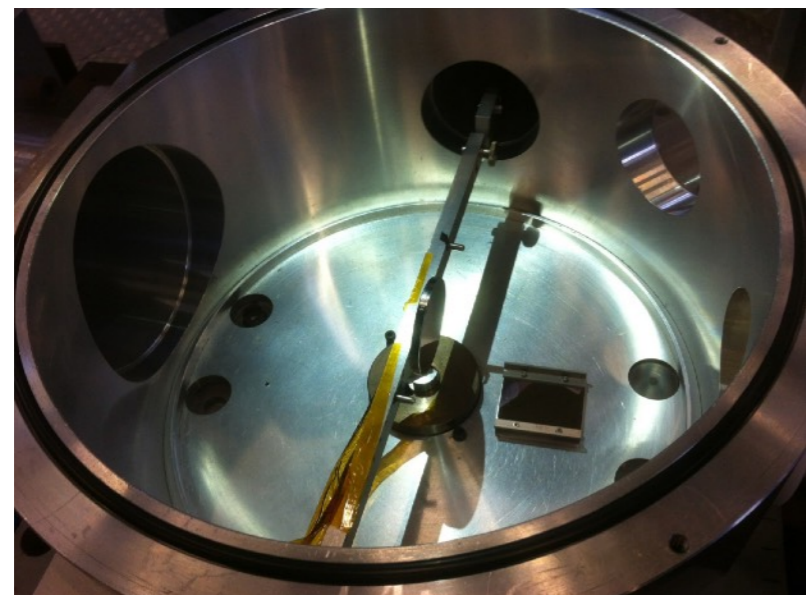
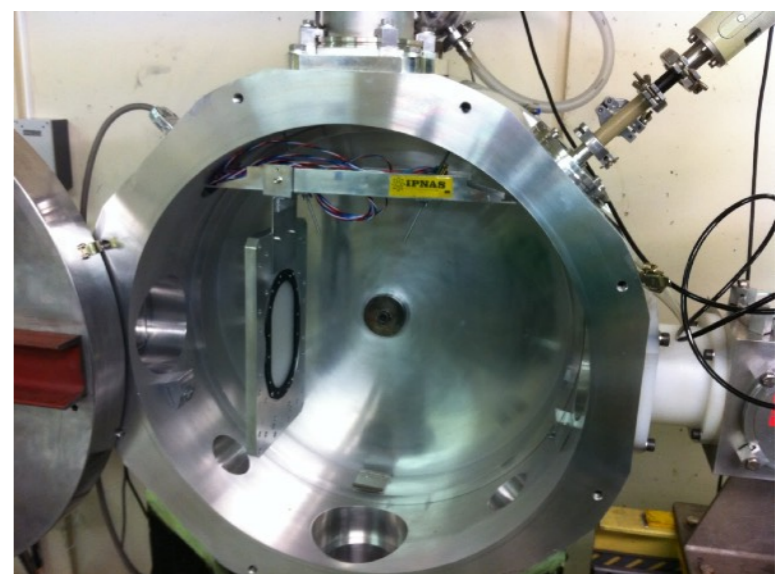
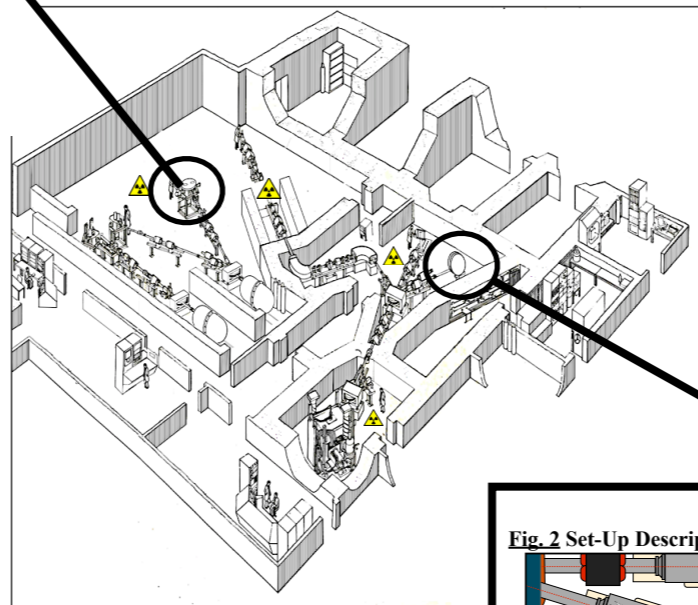


Fig. 2 Set-Up Description

Camera CCD 2	no beam
Thermocontrolled Sample Holder	Aluminium Bulk + 2 PT100
Liquid Nitrogen Cold Trap	Copper Bulk / no beam
Internal Faraday Cup Copper Coated	Thick Copper Coating
Thermal Bridge LNCT-SH	Copper, no beam

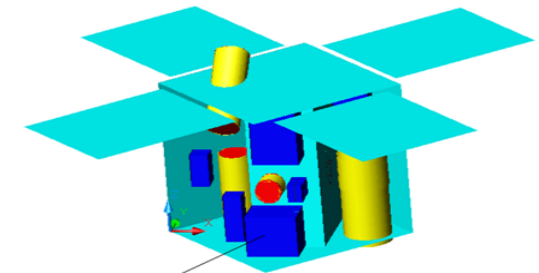
Secondary Electron Suppressor	Stainless Steel Tube at +80V
Beam Defining Collimator	Aluminium
Beam Homogenizer	Aluminium Foil 20 microns
Retractable Luminescent Alumina	Al ₂ O ₃ : Cr, no beam
Camera CCD 1	no beam

Primary Pump	Agilent Tech. SH110 DryScrollPump
Turbo Molecular Secondary Pump	Pfeiffer HiPace 80
Full Range Vacuum Gauge	Pfeiffer Compact Full Range
Remise à l'air	Zeolite Molecular Sieve & Dry Nitrogen

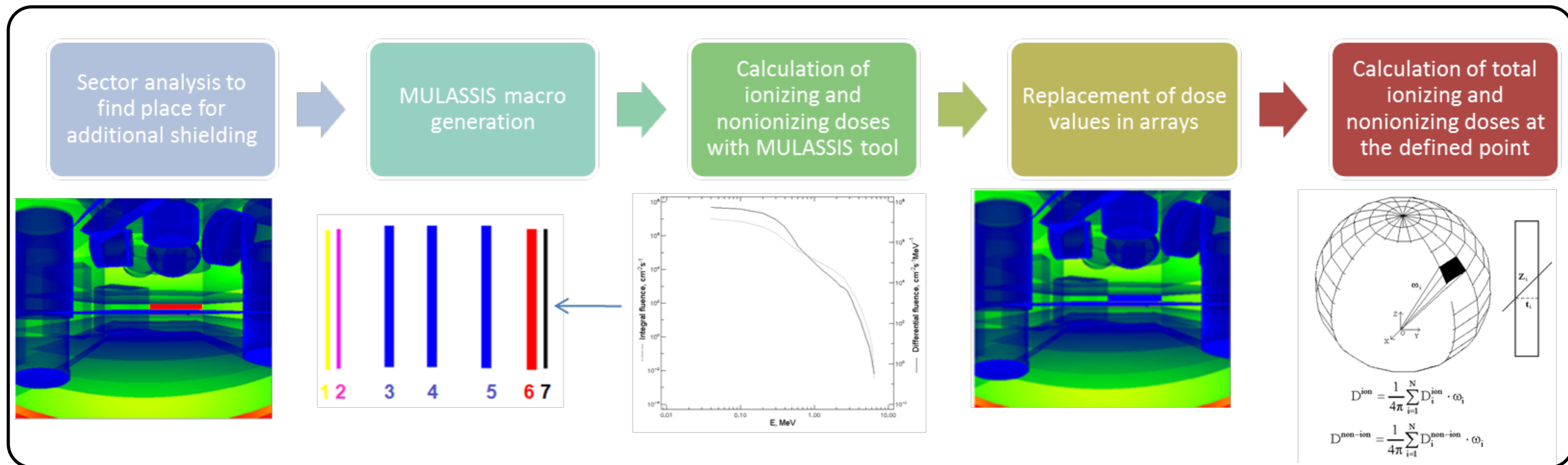
Description : CSL Beam Line, CSL Vacuum Irradiation Chamber, Vacuum Temperature & Fluence control

VDG «CSL Beam Line»
 H+ @ 595 keV / 100 keV on target(S4UVN)
 use of Al foil as beam homogeneizer
 «CSL VDG Chamber»
 LN Cold Trap & I, T, P control and log (S4UVN)

- **Spatial hardware:** use of Al as **radiation shielding** as well as container
- Replace Al by **composite materials** with better weight/mechanical properties
- Graphite/epoxy composites:
 - at equal mass, composites attenuate 30 à 40% less than Al
 - at equal attenuation, composites are 30 to 40% thicker than Al
- Optimization
 - **Z to add** to attain similar shielding with minimal mass
 - **Quantity** to add;
 - **Position** in composite/enveloppe



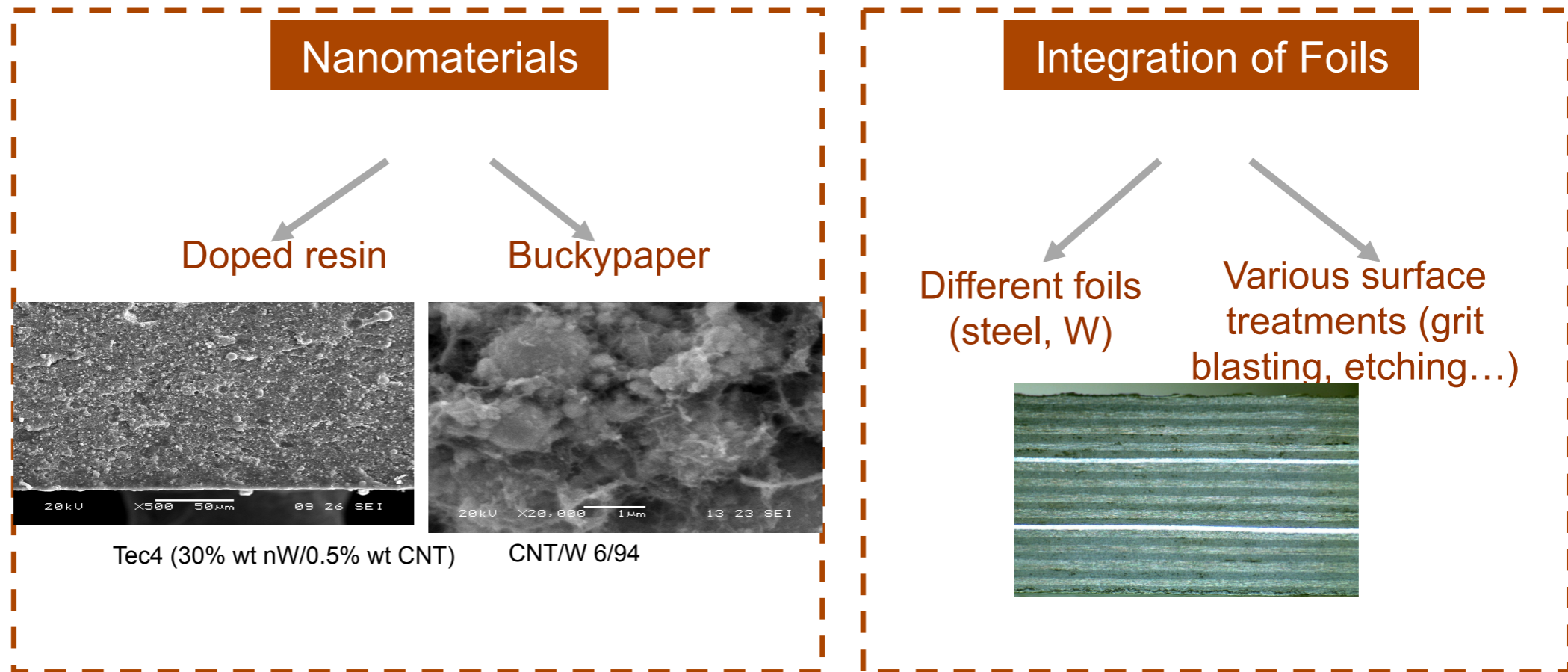
Block of onboard radioelectronic equipment been used for example of calculations





Materials for composite shieldings

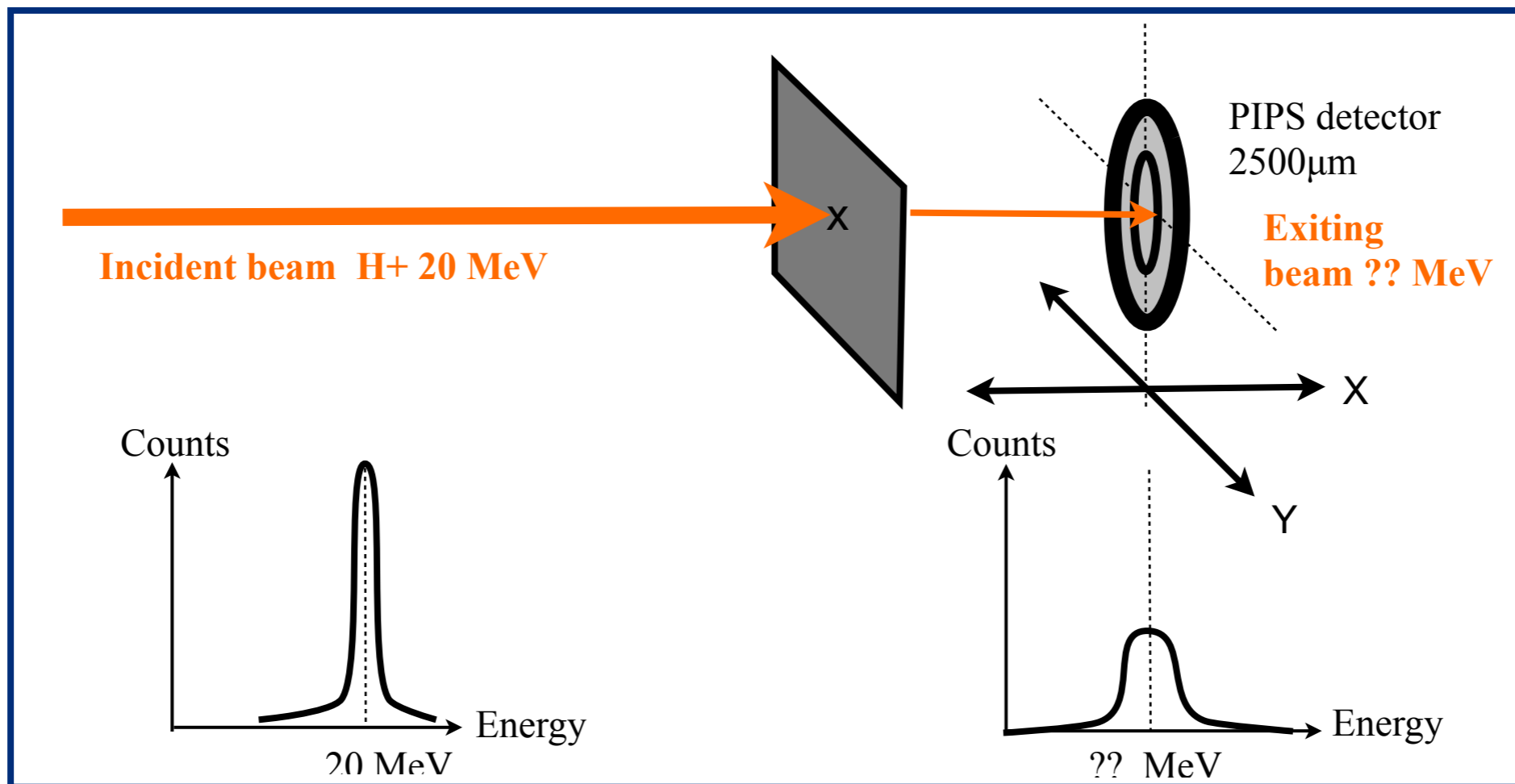
MTM44 epoxy from ACG
fibres M40J

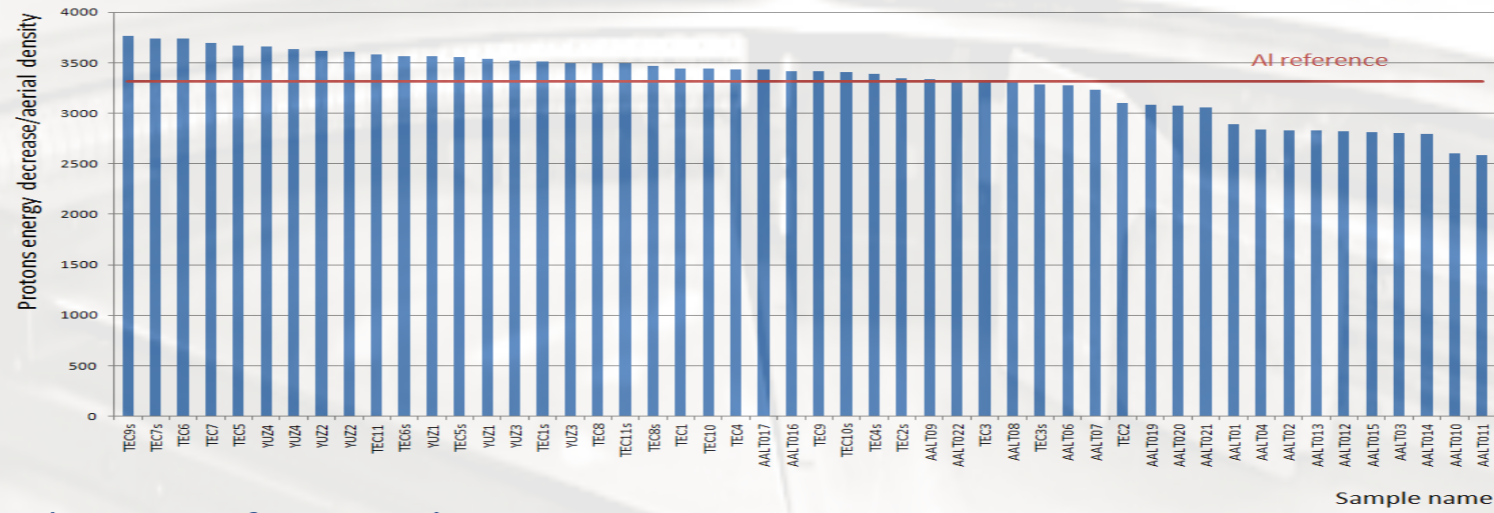




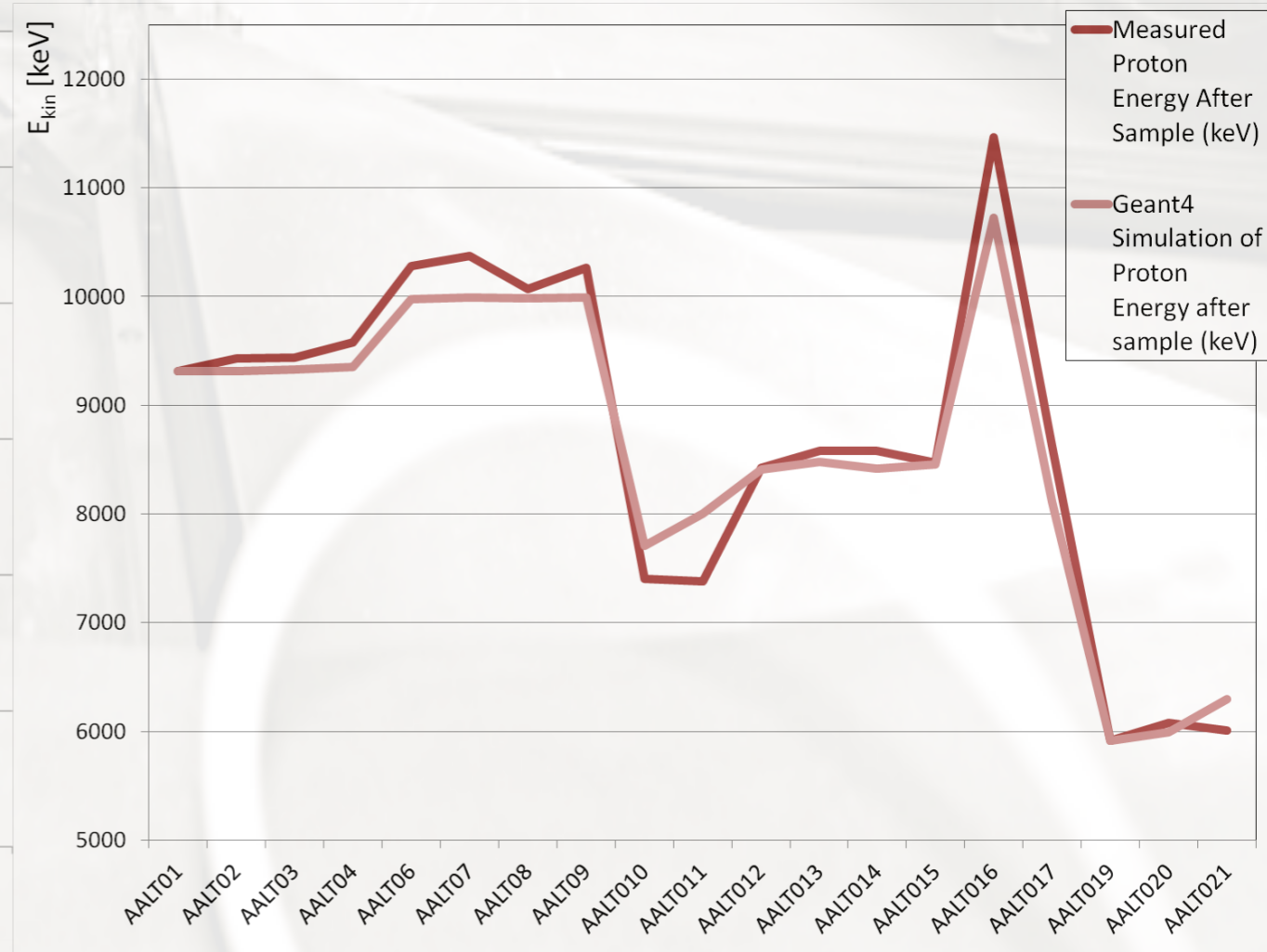
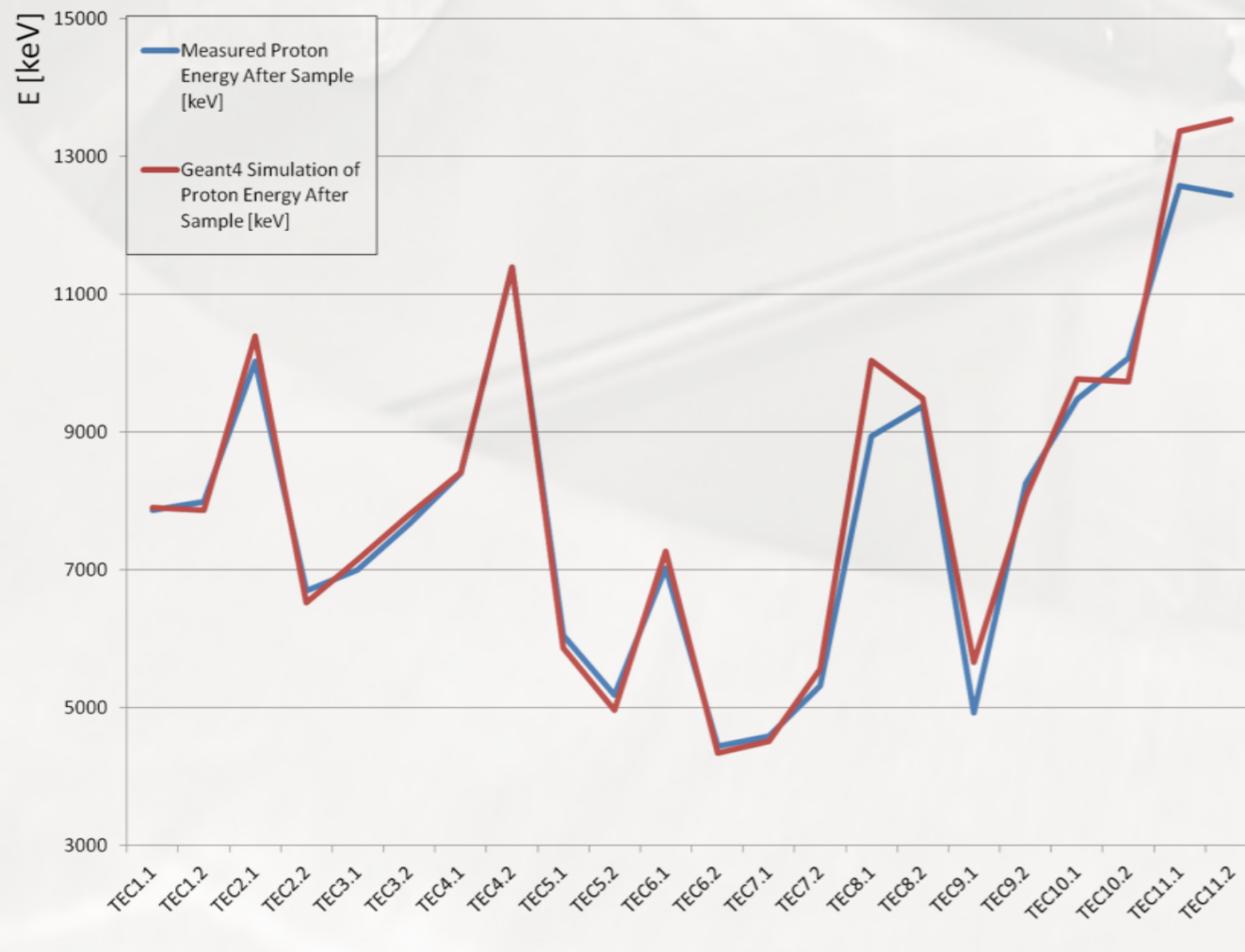
Reference	Details
TEC1	Doped resin with 2,21%CNT
TEC2	Doped resin with 88%W
TEC3	Doped resin with 63%W
TEC4	Doped resin with 30%W/0,5%CNT
TEC9	Buckypaper 100%CNT
TEC8	Buckypaper 76,60%CNT/23,4%W
TEC7	Buckypaper 50%CNT/50%W
TEC6	Buckypaper 24%CNT/76%W
TEC5	Buckypaper 6%CNT/94%W
TEC11	0,Buckypaper 76,60%CNT/23,4%W,90,Buckypaper 76,60%CNT/23,4%W,0
TEC10	Impregnated Buckypaper 30%W
Aalto-1 & Aalto-6	2 prepreg layers + 0.05 mm Tungsten + 4 prepreg layers
Aalto-2 & Aalto-7	3 prepreg layers + 0.05 mm Tungsten + 3 prepreg layers
Aalto-3 & Aalto-8	4 prepreg layers + 0.05 mm Tungsten + 2 prepreg layers
Aalto-4 & Aalto-9	5 prepreg layers + 0.05 mm Tungsten + 1 prepreg layer
Aalto-10:	3 prepreg layers + 0.05 mm Tungsten + 1 prepreg layer + 0.05 mm Tungsten + 2 prepreg layers
Aalto-11:	4 prepreg layers + 0.05 mm Tungsten + 1 prepreg layer + 0.05 mm Tungsten + 1 prepreg layer
Aalto-12:	3 prepreg layers + 0.05 mm steel + 1 prepreg layer + 0.05 mm Tungsten + 2 prepreg layers
Aalto-13:	4 prepreg layers + 0.05 mm steel + 1 prepreg layer + 0.05 mm Tungsten + 1 prepreg layer
Aalto-14:	2 prepreg layers + 0.05 mm steel + 2 prepreg layers + 0.05 mm Tungsten + 2 prepreg layers
Aalto-15:	3 prepreg layers + 0.05 mm steel + 2 prepreg layers + 0.05 mm Tungsten + 1 prepreg layers
Aalto-18:	8 prepreg layers + Gadolinium paint
Aalto-19:	4 prepreg layers + 0.05 mm Tungsten + 4 prepreg layers
Aalto-20:	6 prepreg layers + 0.05 mm Tungsten + 2 prepreg layers
Aalto-21:	7 prepreg layers + 0.05 mm Tungsten + 1 prepreg layer
Aalto-16:	6 prepreg layers (reference)
Aalto-17:	8 prepreg layers (reference)
Aalto-22:	2 mm Al-2024-T3 (reference)

Particles	Energy	Facility	Deliverable
H ⁺	20 MeV	IPNAS ULg Cyclootron AVF	<ul style="list-style-type: none"> Fast acquisition H⁺ energy spectrum with detector placed behind each sample (collimated PIPS, 2500 Microns, Schlumberger) Scan Y & X vs different beam sizes
e ⁻	6 MeV	Medical accelerator	Acquisition of e ⁻ energy spectrum with detector placed behind each sample
γ	⁶⁰ Co	⁶⁰ Co ULN	Acquisition of intensity with detector placed behind each sample



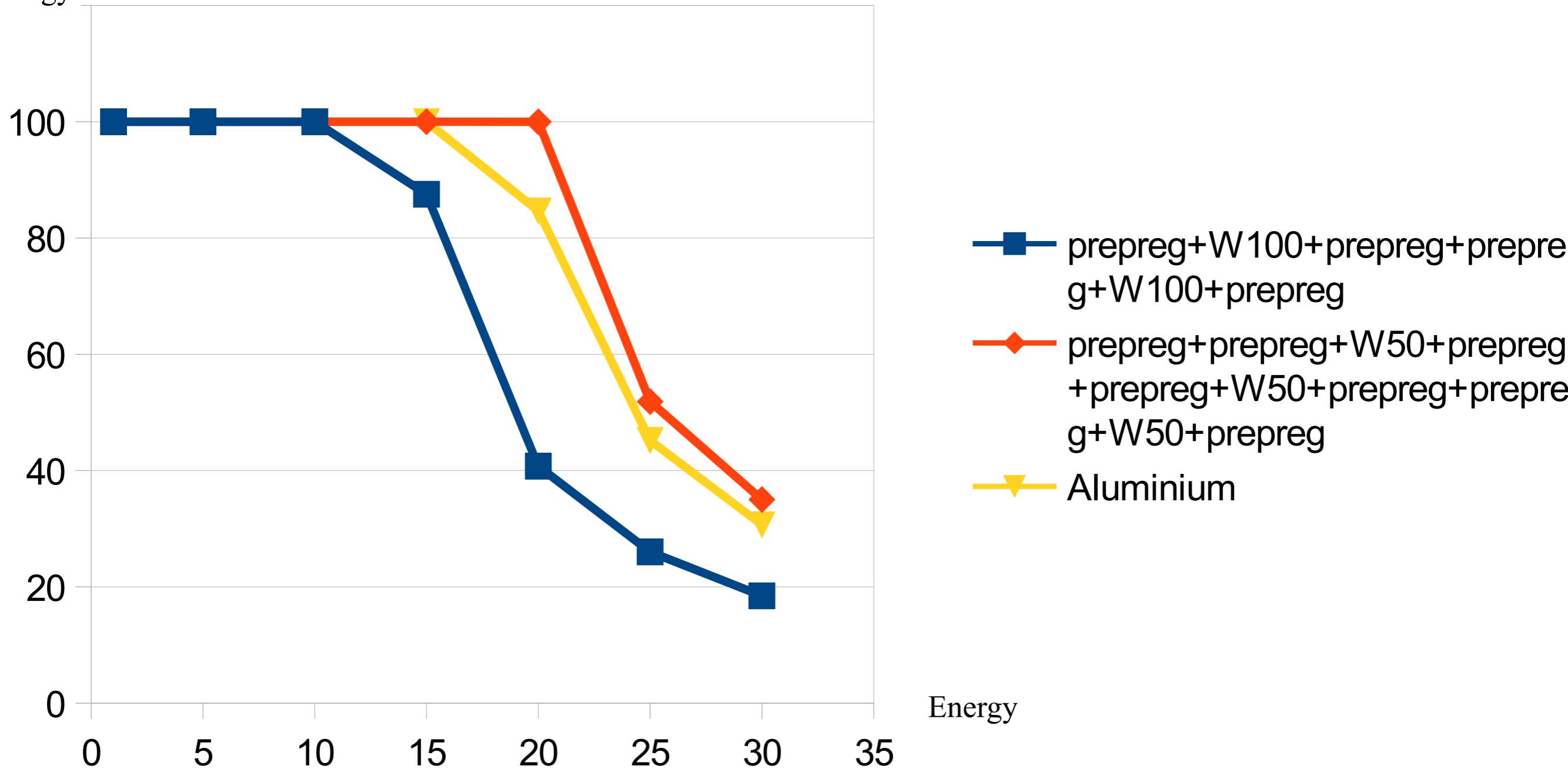


Correlation between Geant4 Model & Experimental



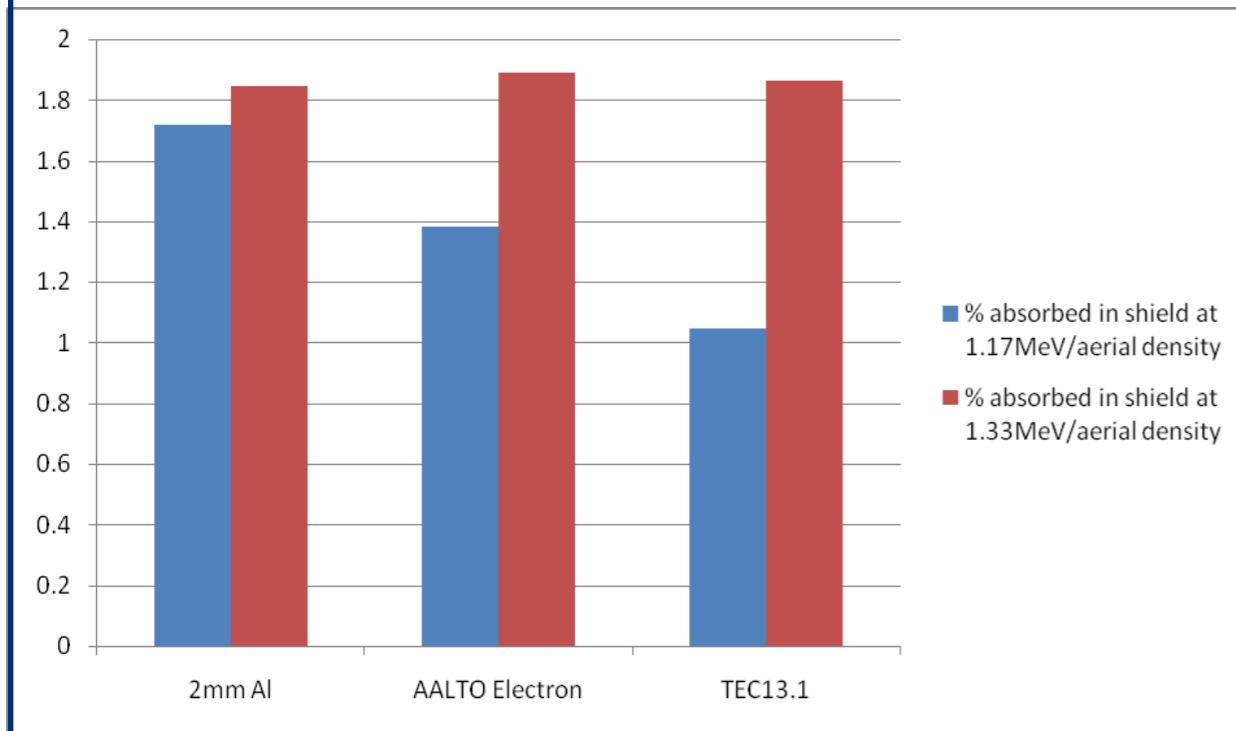
Performance predicted by Geant4 for the next test with respect to 2mm Al

Energy loss %



γ

Source : ^{60}Co , measurement of photons spectrum after shielding



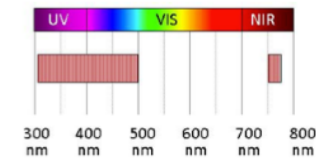
Gamma peak at 1.33 MeV the composite sample is more efficient shielding than Al. However, it is less efficient for the 1.17 MeV peak

e^-

Medical electron accelerator was used to irradiate samples with 6 MeV electrons beam.

	No shield	Al 2mm shield	AALTO shield	TEC13_1 shield
Calculated dose Radfet 1 (cGy)	78.92	18.34	75.74	78.13
Calculated dose Radfet 2 (cGy)	80.52	23.92	74.94	82.91

No more efficient than aluminium. New combinations will be tested in the next phases of the project



programme Copernicus

mission : monitoring de qualité air,
O3, NO2, SO2, HCHO et aerosol optical depth

Orbit : GEO

hyperspectral spectrometer operating with designated spectral bands in the solar reflectance spectrum

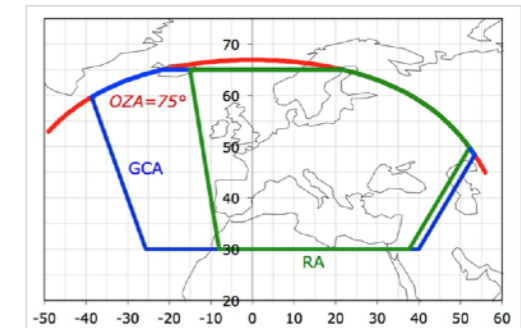
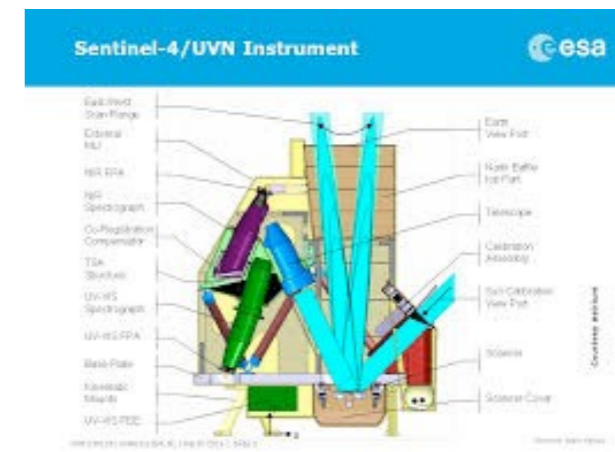
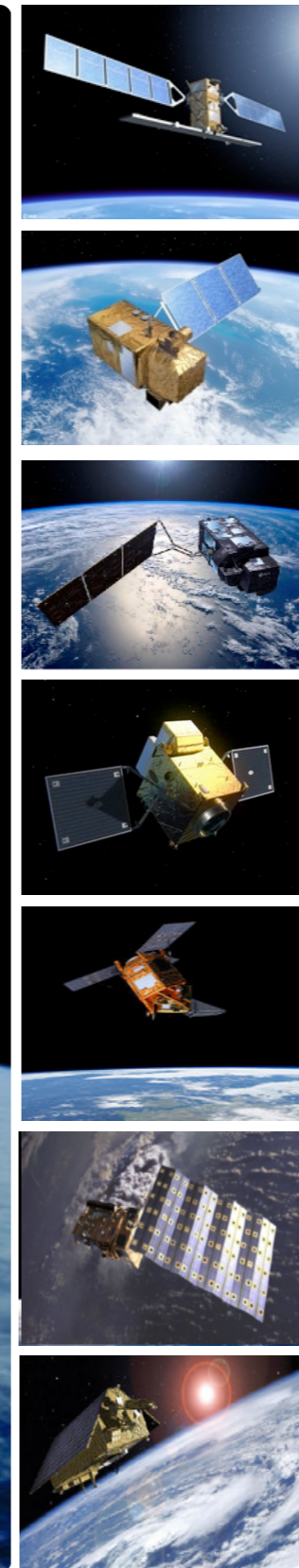


Figure 1: SENTINEL-4 Geographical Coverage Area (GCA) Reference Area (RA) and Observation Zenith Angle (OZA). (Credit: ESA)

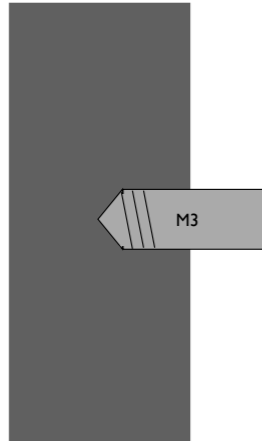
key features spectral range from 305 nm to 500 nm with a spectral resolution of 0.5 nm, and from 750 nm to 775 nm with a spectral resolution of 0.12 nm, in combination with low polarization sensitivity and a high radiometric accuracy.



Fig. 1 Samples Description

Description: Test Sample

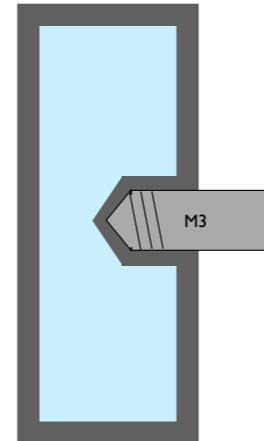
Material	Thickness (μm)
Al	5550



Description: Blank Sample S1_08

Layer name-Role	Material	Thickness
Optical Coating L#1 protective layer	x	x
Optical Coating L#1 protective layer	x	x
Interface - Adhesion layer	x	x
Encapsulating Coating	NiP**	85
Bulk	Be	5500

** (stoichiometry needs to be checked)

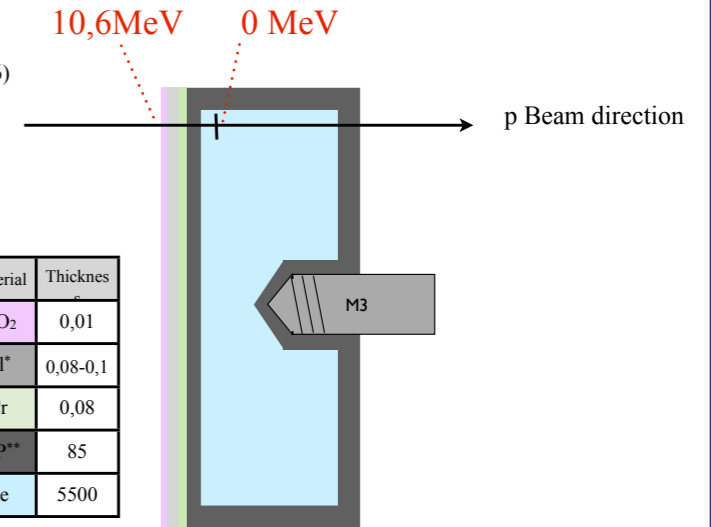


Description: Samples S1_xx (04,11,15,16)

Layer name-Role	Material	Thickness
Optical Coating L#1 protective layer	SiO ₂	0,01
Optical Coating L#1 protective layer	Al*	0,08-0,1
Interface - Adhesion layer	Cr	0,08
Encapsulating Coating	NiP**	85
Bulk	Be	5500

* (Alflex® from OBJ)

** (stoichiometry needs to be checked)



given in the order of a cross-section from the optical surface to the rear surface of the mirror

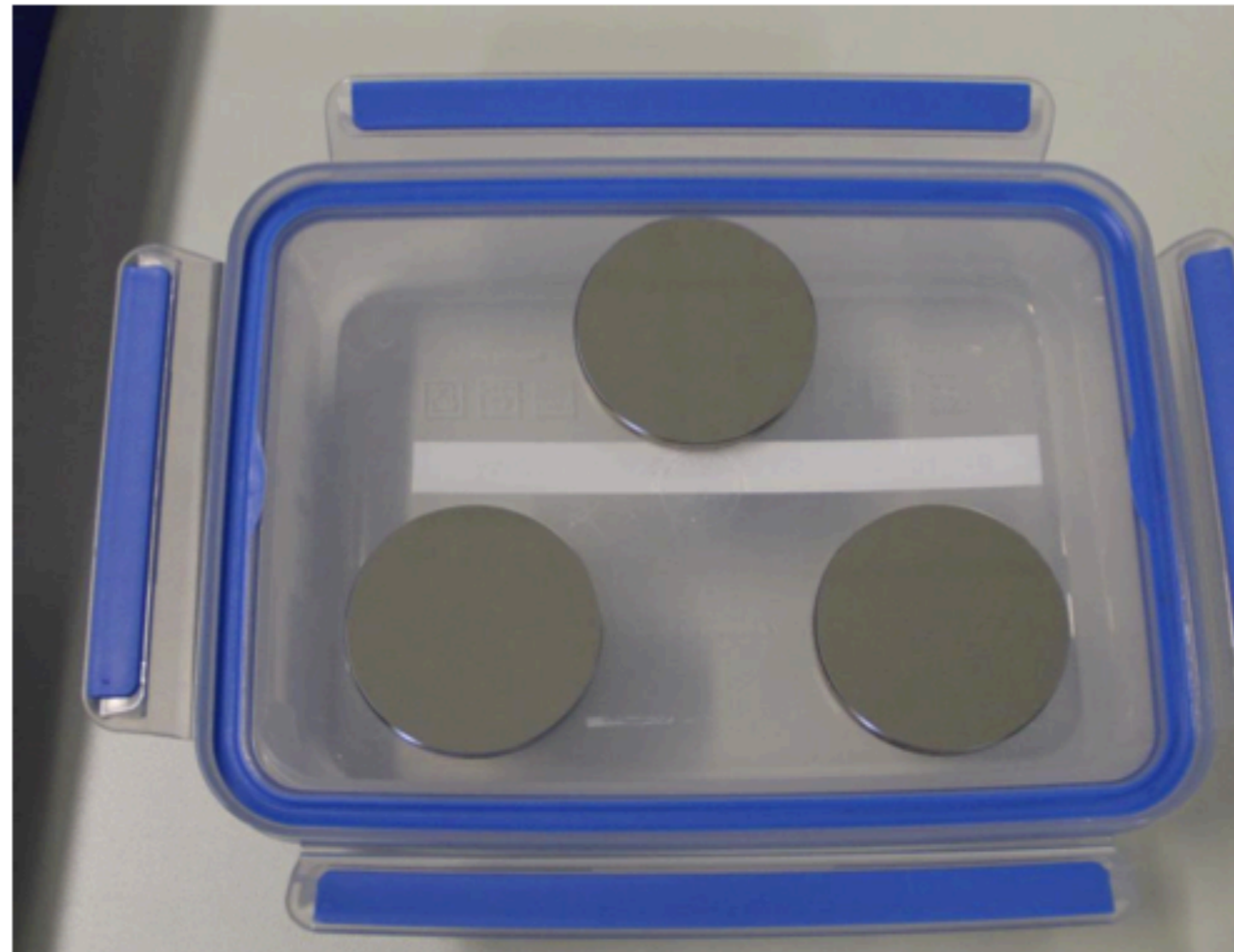


Material/Layer Thickness	Impinging/Outgoing Energy (keV)	Energy Loss (keV)
SiO ₂ /0,010 μm	10600/10599	1
Al/0,08-0,1 μm	10599/10598	1
Cr/0,08 μm	10598/10596	2
NiP/85 μm	10596/8613	1983
Be/5500 μm	8613/0	8613 (All) (Range: 623,4 μm)
NiP/85 μm (rear)	0/0	0

Properties	Phosphorus content (%)		
	Low	Medium	High
Nickel, % (mass)	96 - 99	92 - 95	88 - 91
Phosphorus, % (mass)	1 - 4	5 - 8	9 - 12
Vickers microhardness without heat treatment, HV	650 - 750	500 - 550	450 - 500
Vickers microhardness with heat treatment, HV	1000 - 1050	900 - 950	850 - 900
Melting point, °C	1200	890	870
Density, g/cm ³	8.5 - 8.7	8.1 - 8.3	7.7 - 7.8
Resistivity, μΩ/cm	50	70	90
Resistance to abrasion	Superior	Very good	Very good
Weldability	Good	Regular	Bad

Compound	density (g/cm ³)
SiO ₂	amorphous 2,196
NiP	(m%Ni88P12) 7,7





Vickers microhardness without heat treatment, HV	650 - 750	500 - 550	450 - 500
Vickers microhardness with heat treatment, HV	1000 - 1050	900 - 950	850 - 900
Melting point, °C	1200	890	870
Density, g/cm ³	8.5 - 8.7	8.1 - 8.3	7.7 - 7.8
Resistivity, μΩ/cm	50	70	90
Resistance to abrasion	Superior	Very good	Very good
Weldability	Good	Regular	Bad



Particles	Energy	Facility	Dose	Deliverables
H+	10 MeV	Cyclotron	TID	T control, grounding, P control, contamination , induced activity (gamma spectro)
H+	100keV	VDG	TNID	Grounding, P control, contamination

Table 7-1: TID and TNID figures for Sentinel-4 environment in the coating and in the substrate

Radiation Environment	Irradiation	Fluence [# /cm ²]	TID		TNID	
			coating rad(SiO ₂)	interface rad(Be)	coating MeV/g(Si)	interface MeV/g(Si)
Space	-	-	1.8E+09	2.0E+05	4.8E+14	8.3E+07
On-ground	20 keV electrons	1.3E+16	3.6E+09	0.0E+00	-	-
	10 keV protons	2.9E+14	1.2E+09	-	9.9E+14	-
	14.1 MeV neutron	4.6E+10	-	-	1.7E+08	1.7E+08
	Co-60	2.8E+15	3.27E+04	4.0E+05	-	-
	RDM		2.0	2.0	2.1	2.0

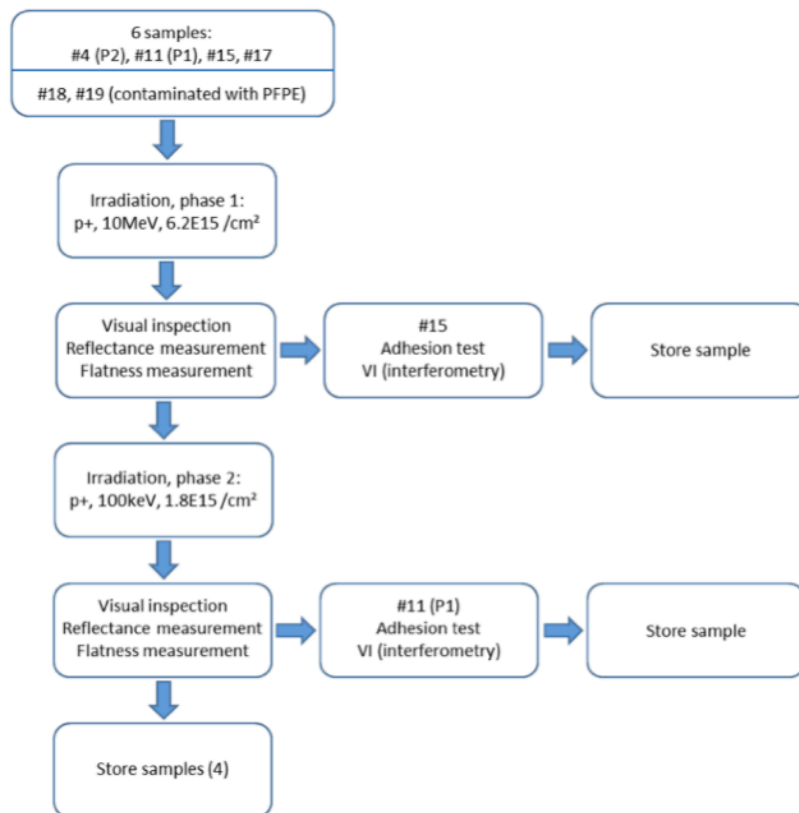
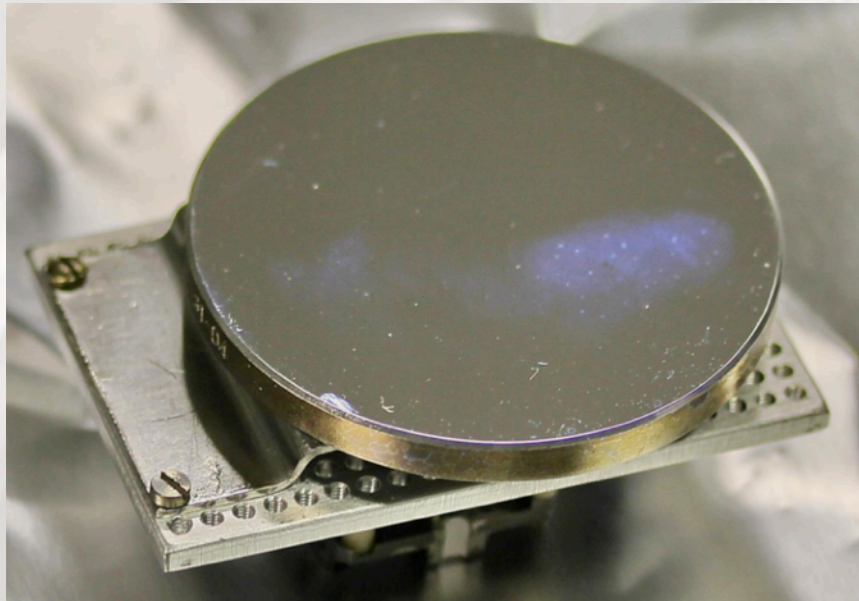


Figure 1: Radiation test flow

Table 7-2: tests parameters

Irradiation step	Irradiation type	Fluence [# /cm ²]
Phase 1	10 MeV protons	6.2E+15
Phase 2	100 keV protons	1.8E+15

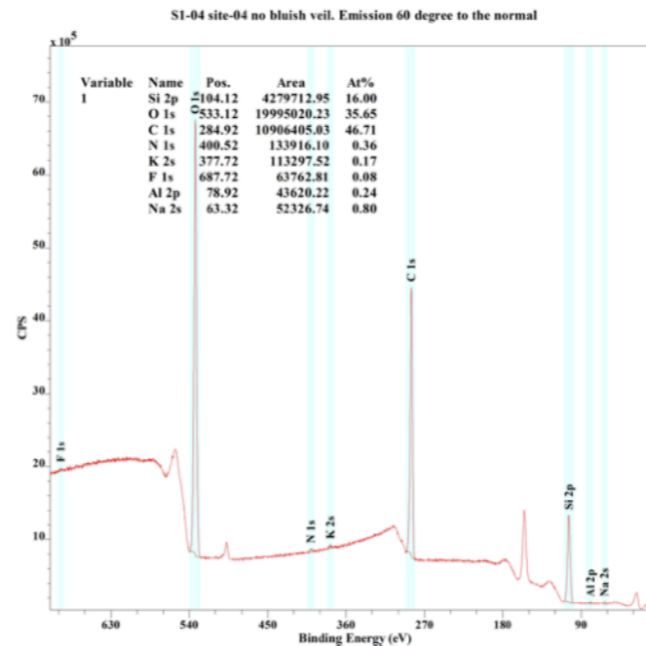
10MeV 6,2 10¹⁵ p/cm²



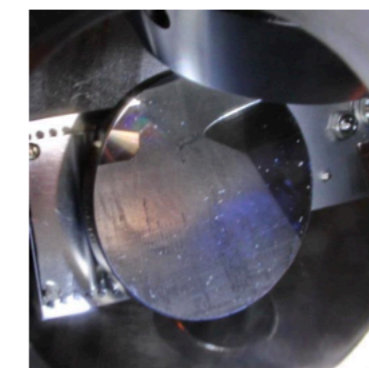
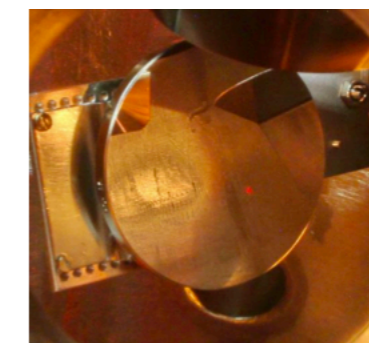
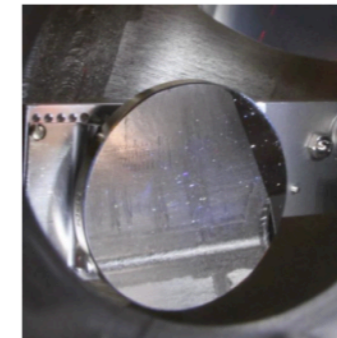
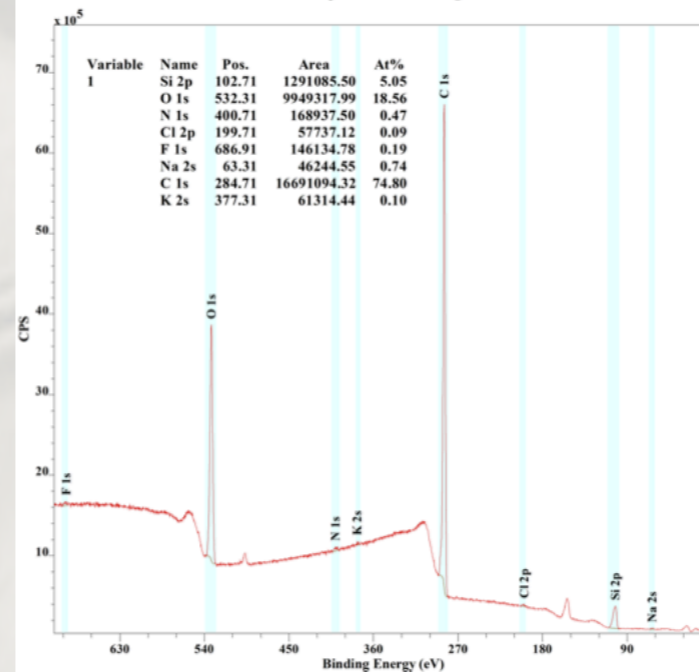
Disqualified mirror set:

- Apparition of a blue voile
- Degradation of optical performances
- Investigation of causes of «**surface contamination ?**»
 - LogFile,
 - FTIR of contamination witness
 - XPS measurements
 - Optical Microscopy and 3D AFM

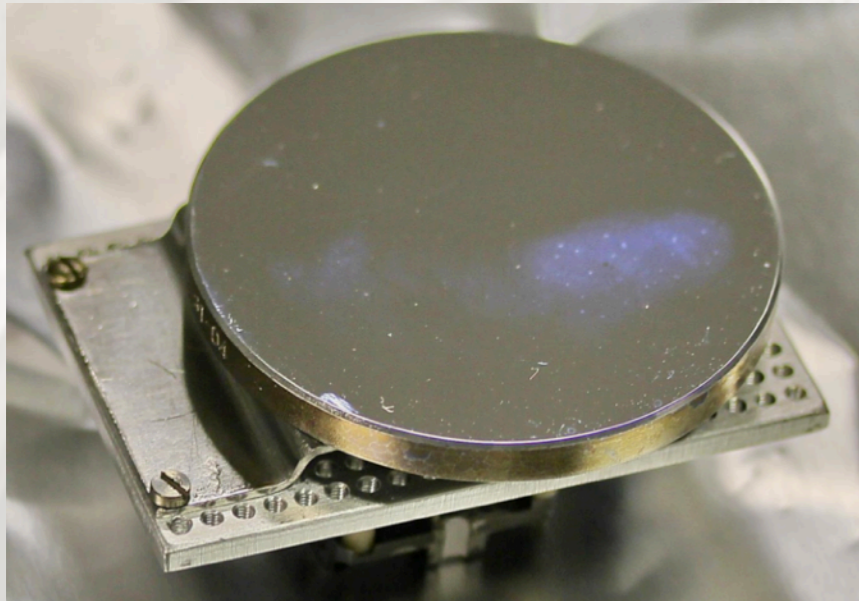
cf. theory



Texte



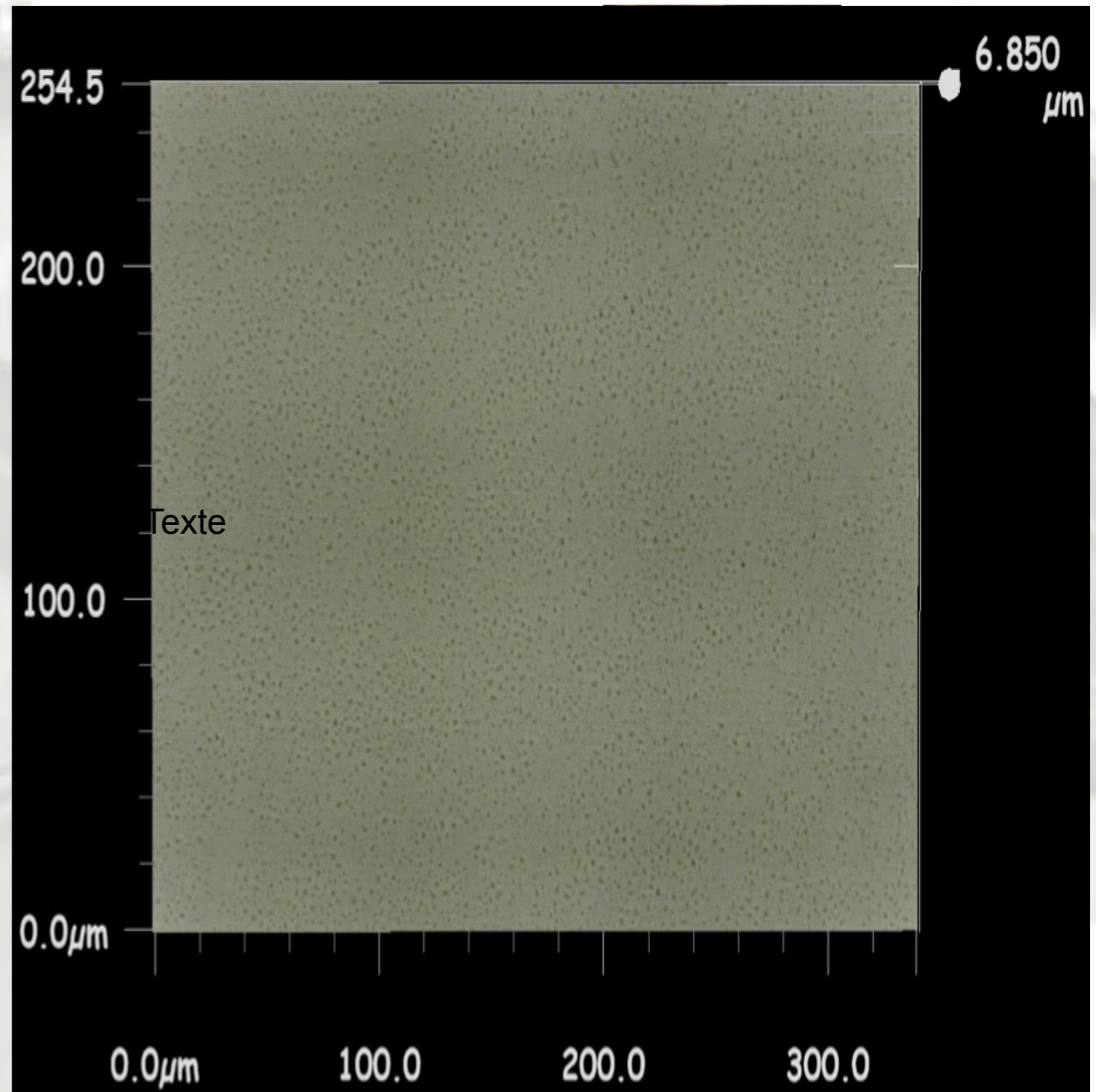
10MeV $6,2 \cdot 10^{15}$ p/cm²



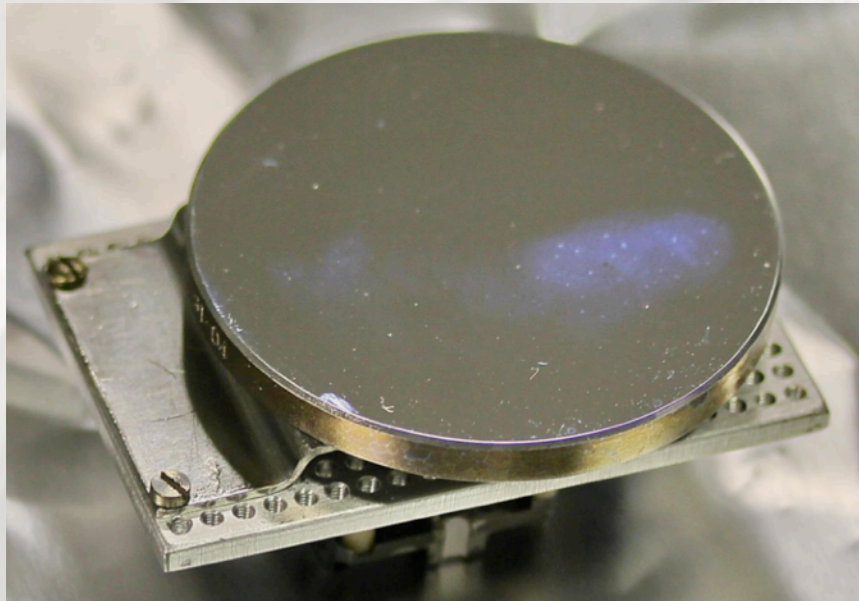
Disqualified mirror set:

- Apparition of a blue voile
- Degradation of optical performances
- Investigation of causes of «**surface contamination ?**»
 - LogFile,
 - FTIR of contamination witness
 - XPS measurements
 - Optical Microscopy and 3D AFM

cf. theory



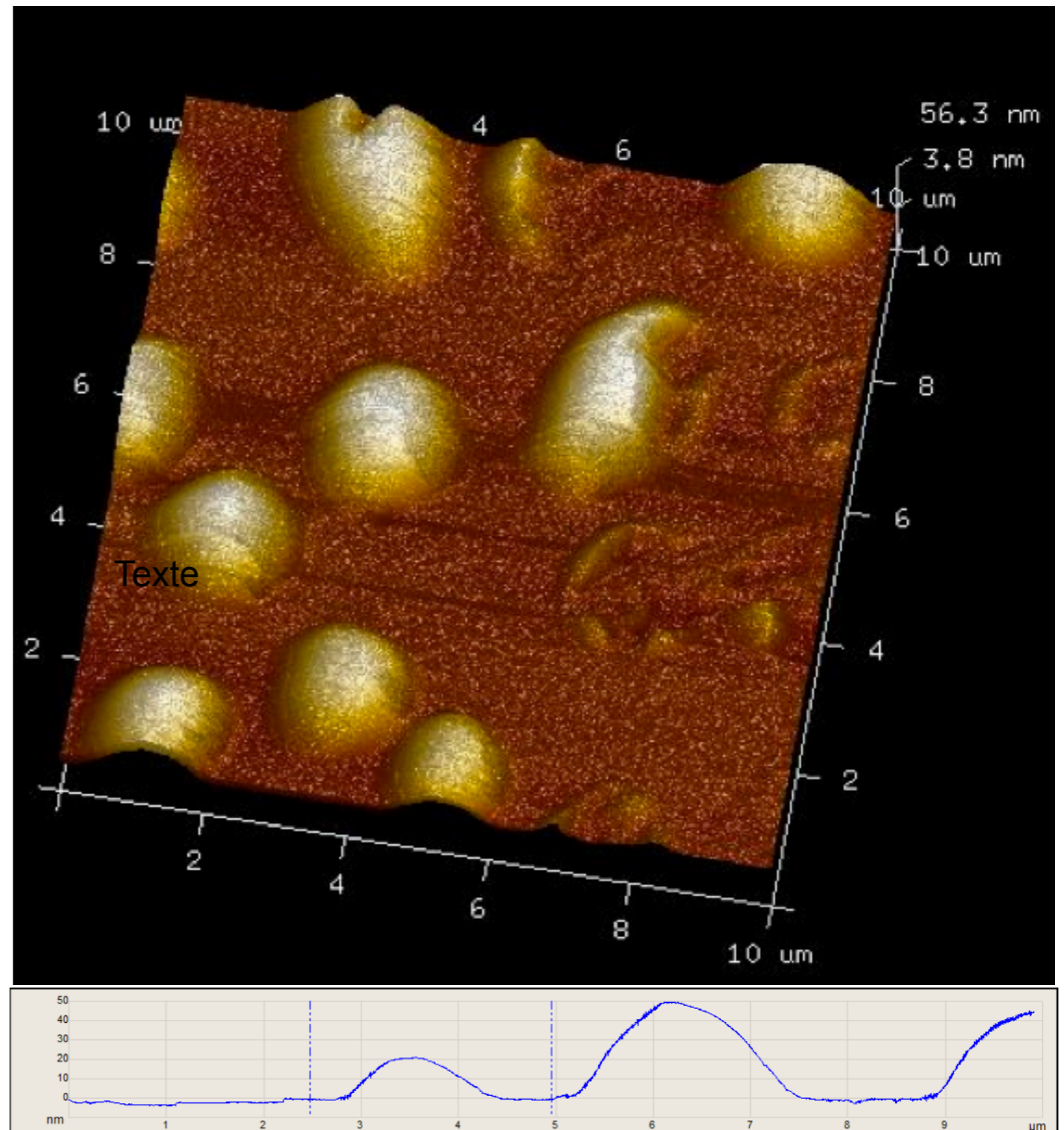
10MeV 6,2 10¹⁵ p/cm²



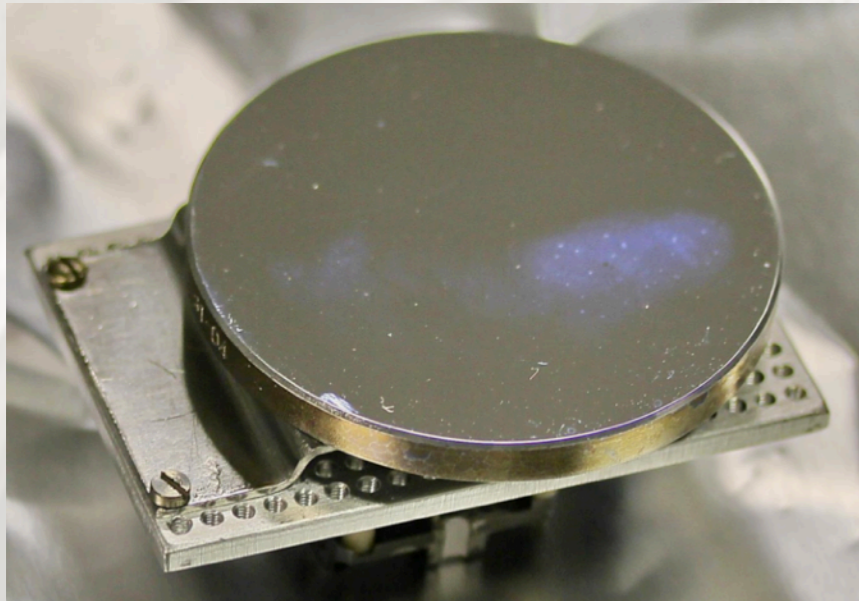
Disqualified mirror set:

- Apparition of a blue voile
- Degradation of optical performances
- Investigation of causes of «**surface contamination ?**»
 - LogFile,
 - FTIR of contamination witness
 - XPS measurements
 - Optical Microscopy and 3D AFM

cf. theory



10MeV 6,2 10¹⁵ p/cm²



Disqualified mirror set:

- Apparition of a blue voile
- Degradation of optical performances
- Investigation of causes of «**surface contamination ?**»
 - LogFile,
 - FTIR of contamination witness
 - XPS measurements
 - Optical Microscopy and 3D AFM

cf. theory

Martynenko et al. «radiation induce bubble»... seuil

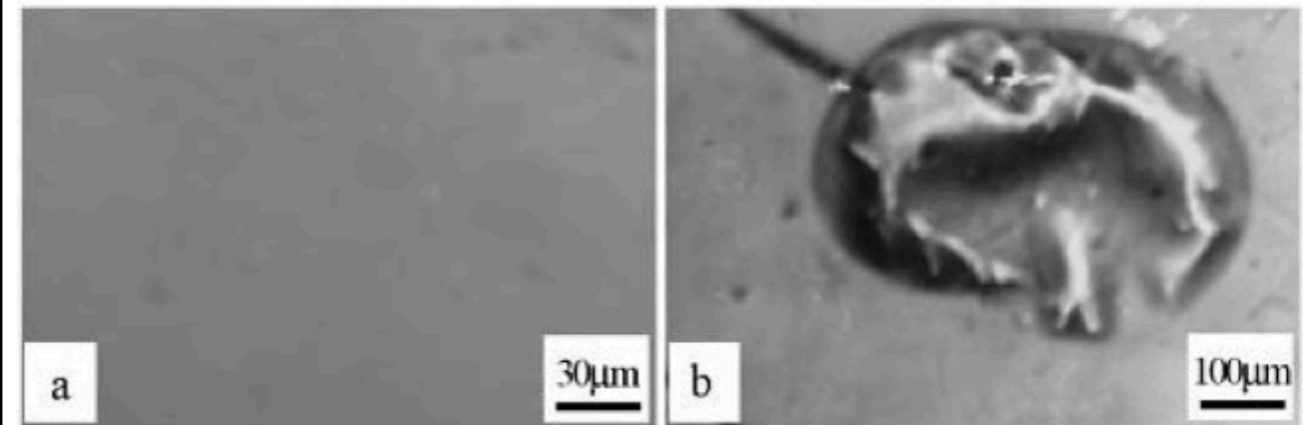


Figure 3. Surface morphologies of the second reflecting mirror before and after proton radiation a - original sample; b - radiated sample, E_p=E_e=130keV, Φ_p=10¹⁶part/cm²

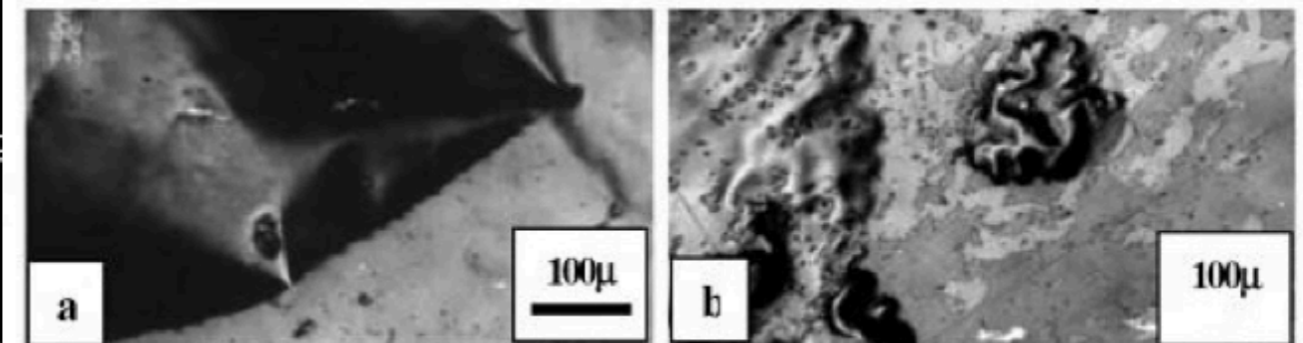
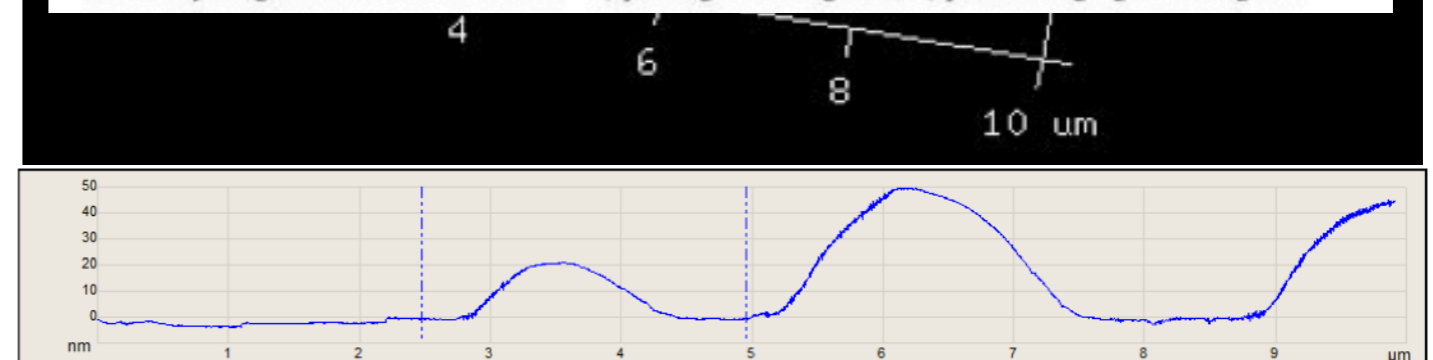


Figure 4. Surface morphologies of the second reflector after proton irradiation and 8 times thermalcycling between 80 and 403 K: a) peeling of the Ag film, b) partial bulging of the Ag film



Particules :	Energie:	Facilité:	Dose :	Délivrables
H+	100keV	VDG	TID	T control, grounding, P control, contamination
H+	100keV	VDG	TNID	T control, grounding, P control, contamination

Table 7-1: TID and TNID figures for Sentinel-4 environment in the coating and in the substrate

Radiation Environment	Irradiation	Fluence [# /cm ²]	TID		TNID	
			coating rad(SiO ₂)	interface rad(Be)	coating MeV/g(Si)	interface MeV/g(Si)
Space	-	-	1.8E+09	2.0E+05	4.8E+14	8.3E+07
γ-ground	20 keV electrons	1.3E+16	3.6E+09	0.0E+00	-	-
	10 keV protons	2.9E+14	1.2E+09	-	9.9E+14	-
	14.1 MeV neutron	4.6E+10	-	-	1.7E+08	1.7E+08
	Co-60	2.8E+15	3.27E+04	4.0E+05	-	-
	RDM		2.0	2.0	2.1	2.0

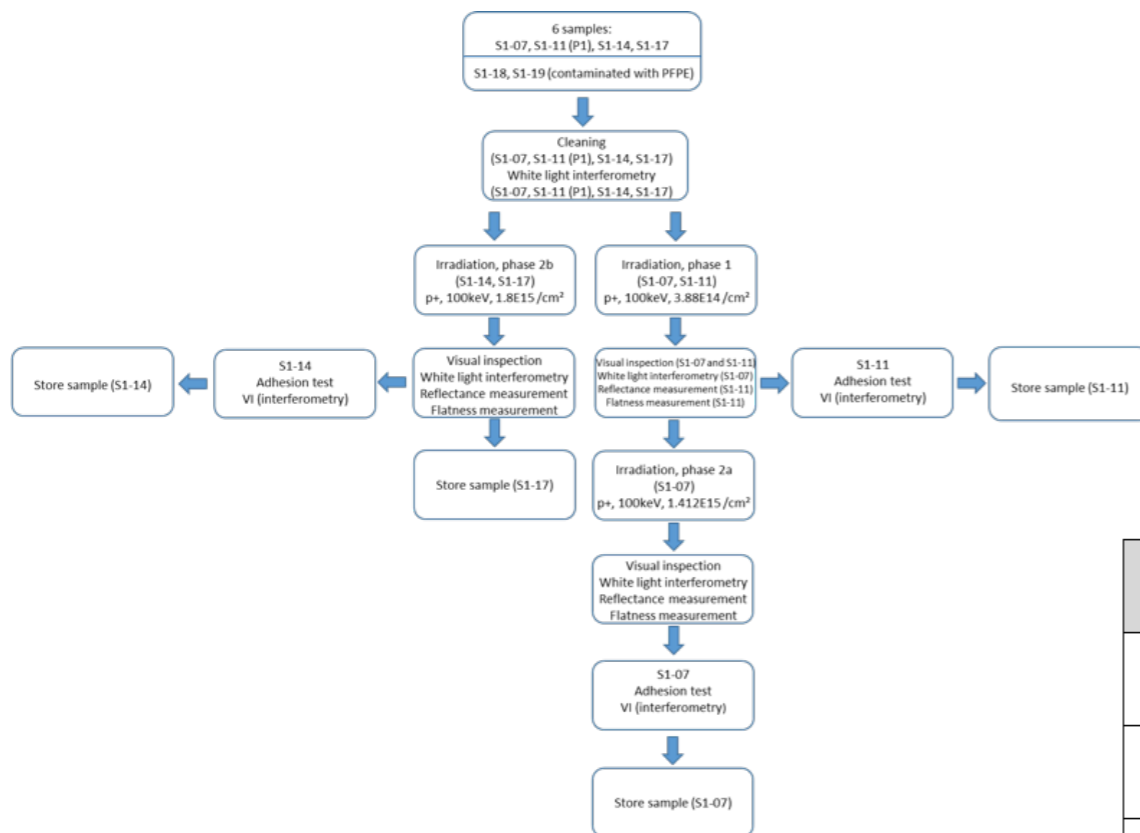
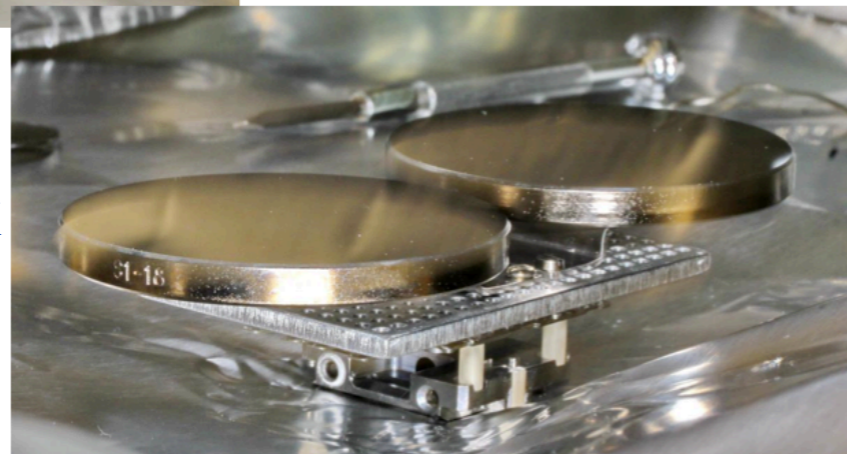
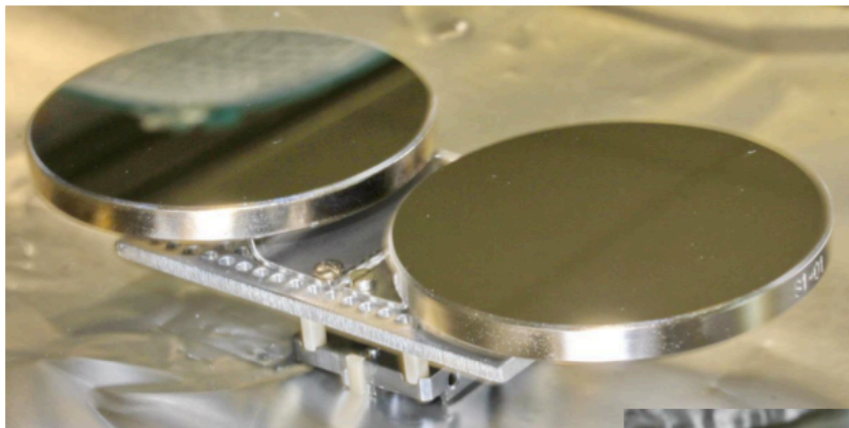


Table 7-2: tests parameters

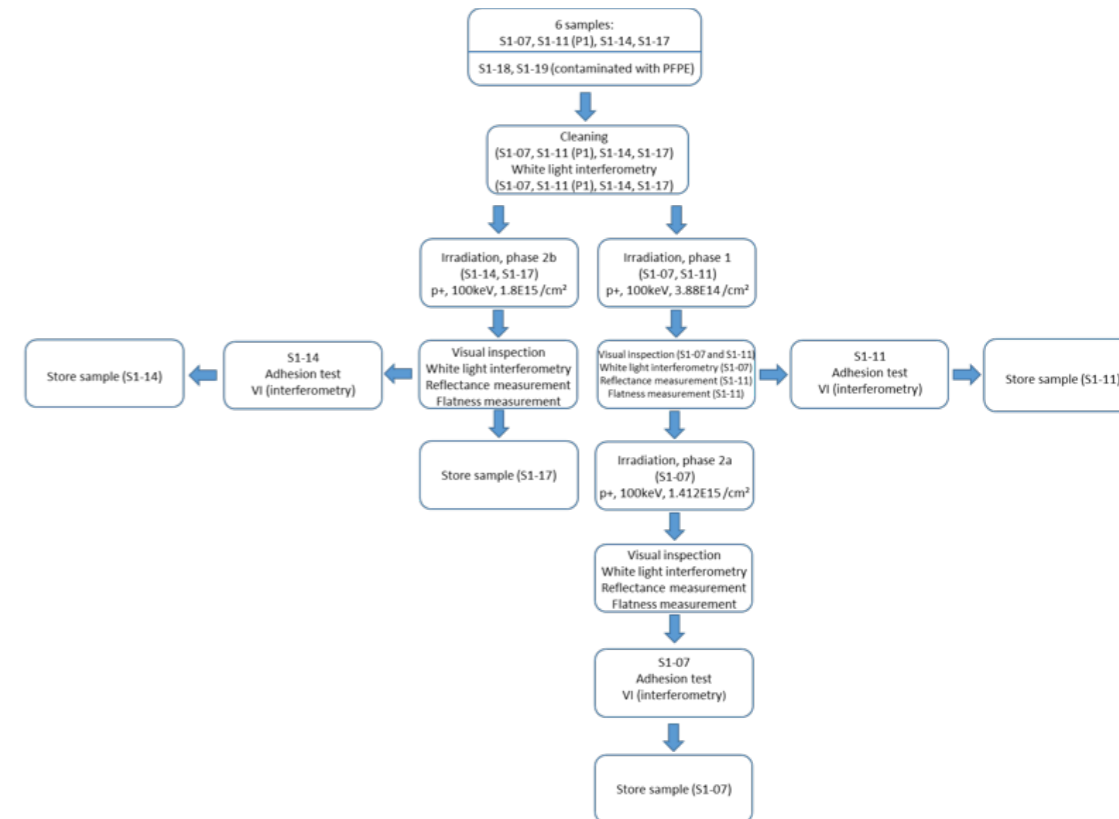
Irradiation step	Irradiation type	Fluence [# /cm ²]	Flux [# /cm ² /s]	Duration	Samples #
Phase 1 (TID)	100 protons keV	3.88E+14	4.49E+9	24h	#7, #11
Phase 2a (TNID)	100 protons keV	1.412E+15	4.49E+9	86h	#7
Phase 2b (TNID)	100 protons keV	1.8E+15	4.49E+9	111h	#14, #17

100 keV All doses



visual inspection **OK**
 optical performance test **OK**
 I,P,T LogFile analysis, **OK**
 FTIR measurements **OK**

Test successful



- New accelerator: use of proton therapy center to get 20-230 MeV protons (2019)







Thank you for your attention





Fundings :



Acknowledgements:

-  K. Fleury-Frenette
- B. Marquet
- J. Y. Plesseria
- J. H. Lecat
- L. Rossi
- A. Carapelle
- O. Dubreuil
- T. Jacquemart
- A. Marchal
- T. Schoonjans

